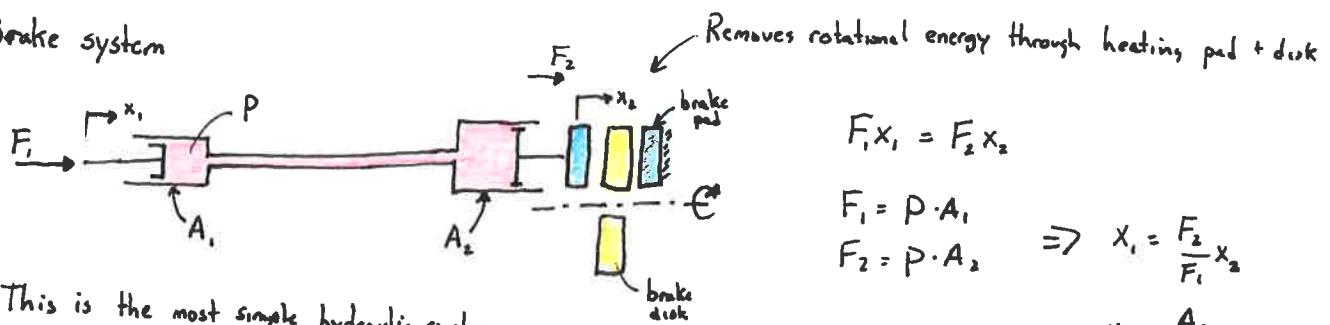

AEM 617
Hydraulic Systems
part 1

Consider the challenges of FCS (Flight Control Systems).

- Large Forces (i.e. $F_h = C_h g S_c$)
- Maintain static position with high precision in presence of large forces
- Rapid control deflections (i.e. high power since $F \cdot v = \text{power}$)
- Widely spaced control surfaces (e.g. ailerons are many feet apart ... same with elevators)
- Control inputs are far from surfaces and need high gain (force ratio)
- Near continuous operation
- Weight constraints
- Redundancy and reliability

A hydraulic system meets these requirements through the use of a pressurized fluid acting on a piston area (or on a motor).

Ex: Brake system



This is the most simple hydraulic system. Provides a force gain of A_2/A_1 , necessary to give enough force on the brake pads to stop rotation relatively quickly.

$$\text{Energy of aircraft} \equiv E = \frac{1}{2} m V^2 \Rightarrow \frac{d}{dt}(E) = m V \frac{dV}{dt}$$

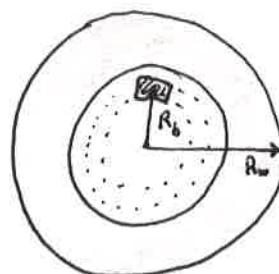
$$\text{Energy balance } \frac{dE}{dt} + P_{\text{brake}} = 0$$

on

$$m V \frac{dV}{dt} + \mu \frac{V}{R_w} R_b F_N = 0 \Rightarrow \frac{dV}{dt} + \underbrace{\mu \frac{R_b}{R_w} \frac{F_N}{m}}_{} = 0$$

Kinetic Friction

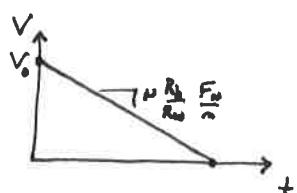
$\mu \approx 0.3 - 0.4$ for normal brakes
 ≈ 0.6 for high performance brakes.



Kinetic friction
 $F_f = \mu F_N \cdot N_b$ # pads

Power loss by brake:

$$P = F_f \cdot V_b = F_f \cdot \omega R_b = \mu \omega R_b F_N$$



Good braking action depends on a brake Kinetic friction

- Ratio of brake to wheel radii (e.g. bicycle brakes)
- Brake normal force

Ex:
Using this brake model, design a brake system for a 2400^{lb} C-172 to stop the aircraft in 500 ft from 50 mph. Assume the tire is 15 in ϕ and the brake pads are 4 inches from the axle.

Given $V = V_0 - kt$ where $K = \mu \frac{R_b}{R_w} \frac{F_N}{m} \cdot N_b$

$$V = \frac{ds}{dt} \Rightarrow \frac{ds}{dt} = V_0 - kt \Rightarrow ds = (V_0 - kt)dt \Rightarrow s = V_0 t - \frac{kt^2}{2}$$

$$s = V_0 t - \frac{kt^2}{2} \quad \text{and} \quad t_0 = \frac{V_0}{k} \Rightarrow s = \frac{V_0^2}{K} - \frac{kV_0^2}{2K^2} t = \frac{V_0^2}{K} - \frac{1}{2} \frac{V_0^2}{K} t$$

$$s = \frac{1}{2} \frac{V_0^2}{K} = \frac{1}{2} V_0^2 \frac{R_w}{R_b} \frac{m}{F_N} \frac{1}{N_b}$$

Mass:

$$2400\text{lb} \Rightarrow 75 \text{ slugs}$$

Vel:

$$50 \text{ mph} \Rightarrow 73 \text{ ft/s}$$

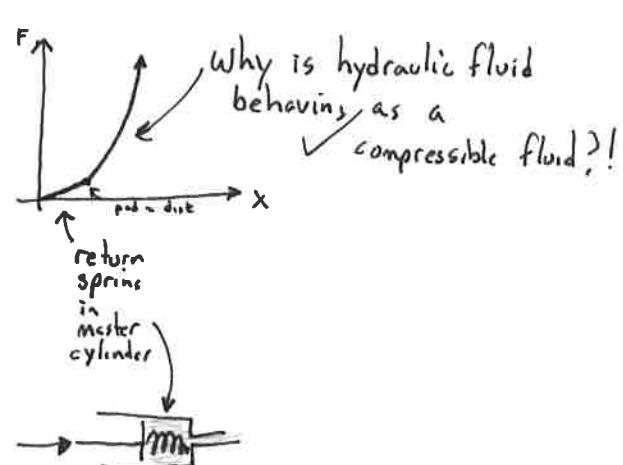
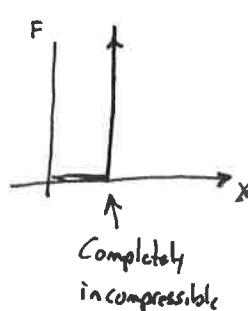
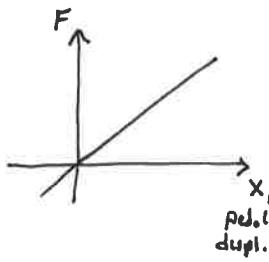
Solution:

$$500 \text{ ft} = \frac{1}{2} \left| \begin{array}{c|c|c|c|c|c|c|c} 73^2 \text{ ft}^2 & 7.5 \text{ in} & 75 \text{ slugs} & F_N & 0.4 & 4 & 16\text{ft} \\ \hline s^2 & 4 \text{ in} & & m & \frac{1}{4} & & \end{array} \right| \text{slug ft}$$

$$F_N = \frac{1}{2} \cdot 73^2 \cdot \frac{7.5}{4.0} \cdot 75 \cdot \frac{1}{0.4} \cdot \frac{1}{4} \cdot \frac{1}{500} = 468 \text{ lbf}$$

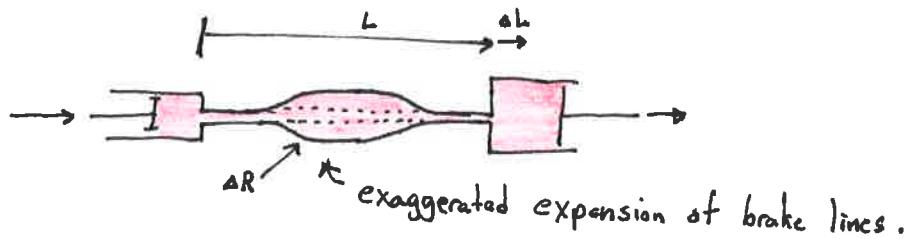
If the pilot can provide 50 lbf to the brake master cylinder, the hydraulic area ratio $\approx 1:10$.

Q: What does the brake force feel like to the pilot?



Hydraulic fluid is only very slightly compressible.

However, the large pressures generated can cause the system to show compressibility because the control volume is not fixed.



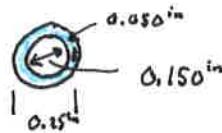
or

Slight lengthening of hydraulic tubes under pressure

Ex: Nylaflo Tubing (Nylon tubing) 10ft ~~mmmmm~~

$\frac{1}{4}$ in OD and 0.050 in walls

$$E \approx 175000 \text{ psi}$$



Hoop stress: (assume thin wall)

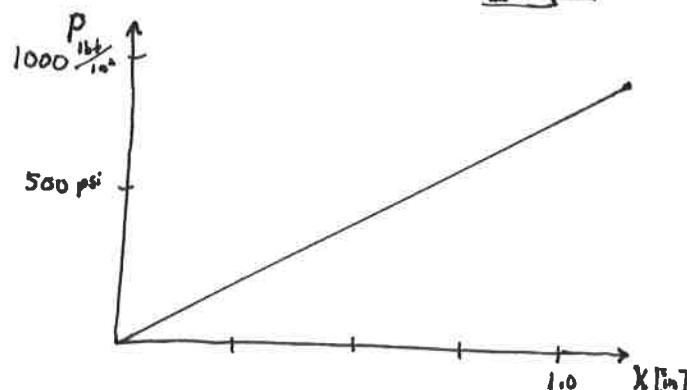
$$\sigma_h = \frac{Pr}{t} \Rightarrow \epsilon = \frac{\sigma}{E} = \frac{Pr}{Et} \Rightarrow \Delta r = \epsilon r = \frac{Pr^2}{Et} \Rightarrow r = r + \Delta r$$

Wall stress:

$$\sigma_x = P \cdot 2\pi r t = P \cdot A \Rightarrow \sigma_x = P \pi r^2 \cdot \frac{1}{2\pi r t} = \frac{Pr}{2t} \quad \text{half hoop....}$$

$$\epsilon = \frac{\sigma}{E} = \frac{Pr}{2Et} \Rightarrow \Delta L = \epsilon L = \frac{PrL}{2Et} \Rightarrow L = L + \Delta L$$

Given a master cylinder area of $\approx 0.2 \text{ in}^2$ (as expected from prev example), the P vs x curve from tubing expansion is

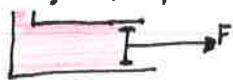


"Soft" Brakes.

1 inch of cylinder travel at 1000 psi with 10 ft of $\frac{1}{4}$ in nylon tubing.

Hydraulic Cylinders

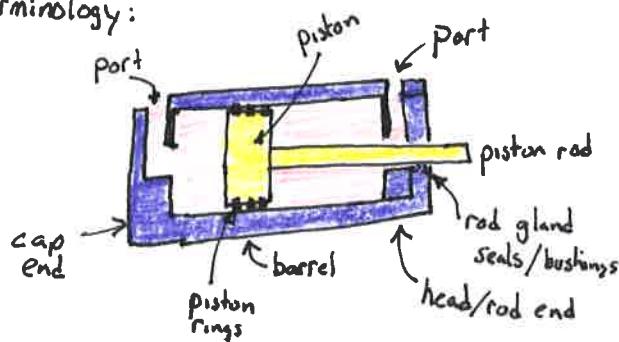
Ram: / Single Acting



Double Acting:

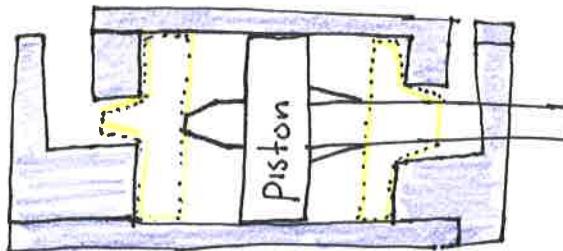


Terminology:

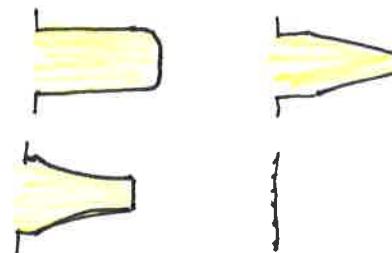


Cushions:

At high flow rates, the piston can slam into the cap end or head end. A tapered spear is added to the piston to reduce the flowrate when the piston approaches the ends.



Cushion spears: Shapes



Biased Forces

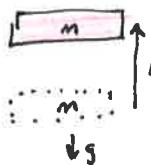
$$\text{Piston} = \text{Area} = \pi R^2 + \text{Area} = \pi R^2 - \pi r^2$$

The force in the head direction can be slightly/moderately less for standard rod types (i.e. non-double rods)

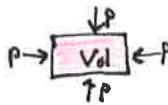
Bernoulli's Eqn.



$$\text{Kinetic energy} \equiv K = \frac{1}{2}mv^2$$



$$\text{Potential Energy} \equiv PE = mgh$$



$$\text{Pressure energy} \equiv Pe = p \cdot Vol$$

Sum energy:

$$K + PE + Pe = E \cdot Vol$$

$$\frac{1}{2}mv^2 + mgh + p \cdot Vol = E \cdot Vol$$

For an arbitrary volume (i.e. divide by Vol)

$$\frac{1}{2} \frac{m}{Vol} V^2 + \frac{m}{Vol} gh + p = E \quad \text{where } \frac{m}{Vol} \text{ is density} \equiv \rho$$

Bernoulli

$$\frac{1}{2} \rho V^2 + \rho gh + p = E$$

For any two pts, E is constant. (assume no losses)

$$\frac{1}{2} \rho (V_1^2 - V_2^2) + \rho g(h_1 - h_2) + (P_1 - P_2) = 0$$

For our application, $\Delta h = h_1 - h_2$ is usually small.

$$\frac{1}{2} \rho (V_1^2 - V_2^2) + (P_1 - P_2) = 0$$

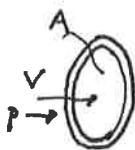
As flow velocity in a hydraulic system increases, pressure decreases

Power

Why is a hydraulic system so effective at power transmission?

$$\text{Power} = F \cdot v$$

\nwarrow
velocity



Force at any point/cross section is $F = \frac{P}{A} A = PA$

$$= P A V$$

\nwarrow
pressure

Volumetric flow rate, \dot{Q}

Ex: How many hp and kW is flow at $p^{[\text{psi}]}$ and $\dot{Q}^{[\frac{\text{ft}^3}{\text{min}}]}$?

$$P_{\text{hp}}^{[\text{hp}]} = P \dot{Q} = \frac{p^{[\text{psi}]} \dot{Q} \left[\frac{\text{ft}^3}{\text{min}} \right] \left[\frac{16 \text{ ft}}{30 \text{ in}} \right] \left[\frac{550 \text{ ft lb}}{550 \text{ in} \cdot 60 \text{ s}} \right] \left[\frac{144 \text{ in}^2}{\text{ft}^2} \right]}{1714} = \frac{p^{[\text{psi}]} \dot{Q} \left[\frac{\text{ft}^3}{\text{min}} \right]}{229.2}$$

$$P_{\text{kW}}^{[\text{hp}]} = \frac{p^{[\text{psi}]} \dot{Q} \left[\frac{\text{gal}}{\text{min}} \right]}{1714}$$

$$P_{\text{kW}}^{[\text{kW}]} = \frac{p^{[\text{bar}]} \dot{Q} \left[\frac{\text{L}}{\text{min}} \right]}{600}$$

Ex: Hydraulic power at 36 gpm at 3500 psi?

$$P_{\text{hp}}^{[\text{hp}]} = \frac{3500 \text{ psi} \cdot 36 \text{ gpm}}{1714} = 73 \text{ hp} \quad \text{ideal \& 100% efficiency}$$

Given a $\frac{1}{2}$ in ID, what is the fluid velocity?

$$\dot{Q} = A v \Rightarrow v = \frac{\dot{Q}}{A} = \frac{36 \text{ gal}}{\pi \cdot 0.5^2 \text{ in}^2} \cdot \frac{231 \text{ in}^3}{\text{gal}} \cdot \frac{\text{in}}{60 \text{ s}} \cdot \frac{\text{ft}}{12 \text{ in}}$$

$$= 58 \frac{\text{ft}}{\text{s}}$$

Hydraulic Pumps

Hydraulic systems in aircraft are typically relatively high pressure (≈ 3000 psi is common). Thus, positive displacement pumps are the norm.

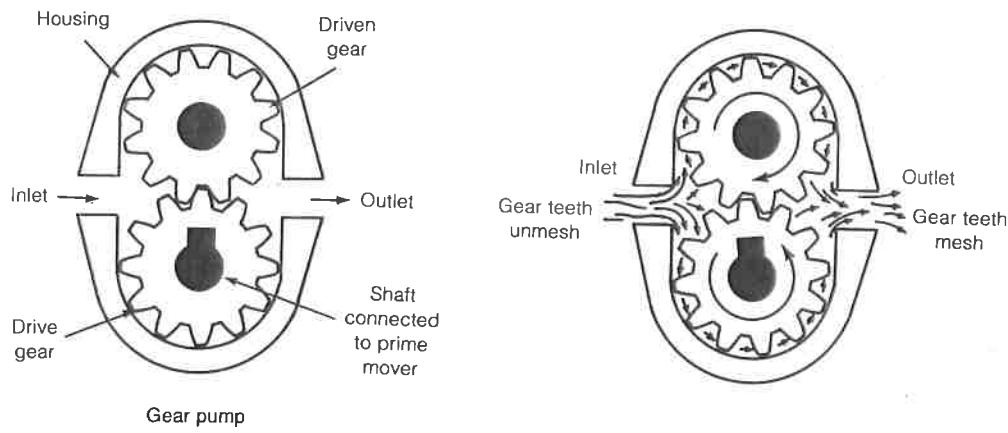


Figure 16.10.1 Gear pump schematic. Source: Parker-Hannifin Corp., Cleveland, OH.

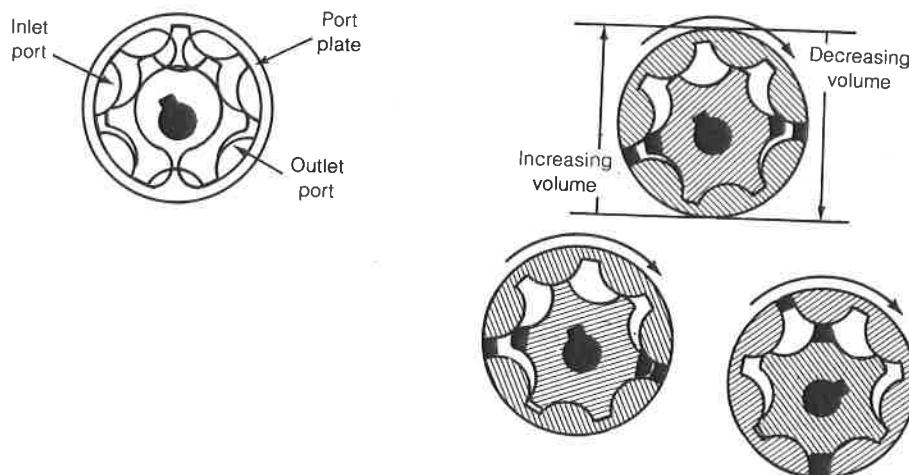


Figure 16.10.2 Schematic of the elements of a gerotor pump. Source: Parker-Hannifin Corp., Cleveland, OH.

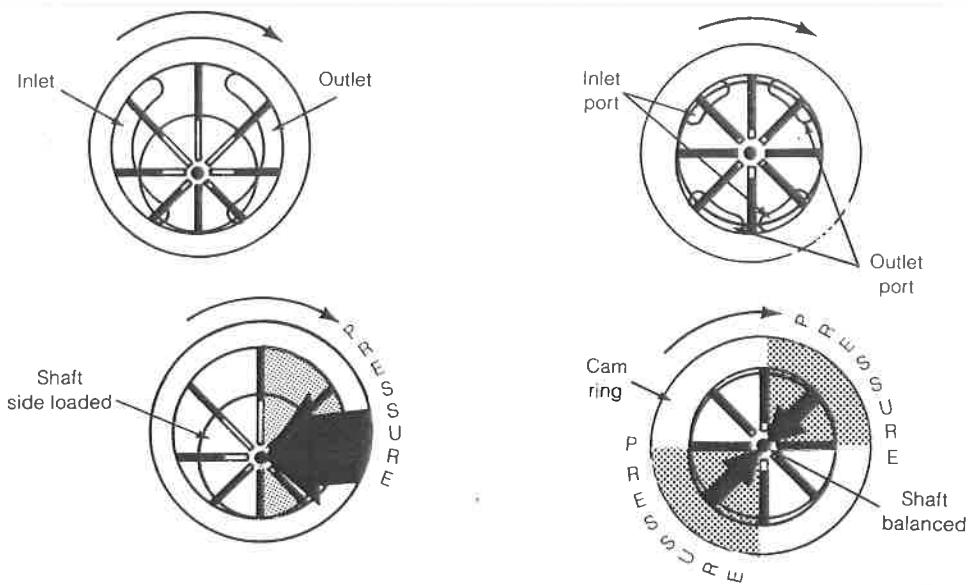


Figure 16.10.4 Cross-sectional schematic of an unbalanced vane pump (*left*), and of a balanced vane pump (*right*). Source: Parker-Hannifin Corp., Cleveland, OH.

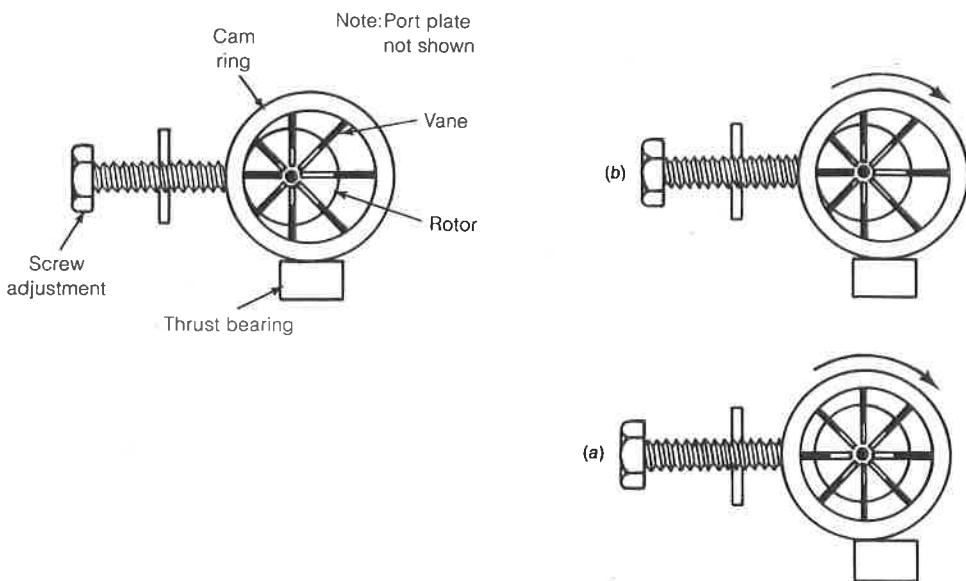
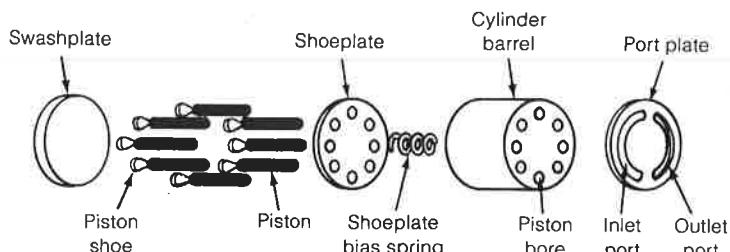


Figure 16.10.5 Cross-sectional schematic of a variable vane pump. Source: Parker-Hannifin Corp., Cleveland, OH.



Overcenter axial piston pump
(Drive shaft not shown)

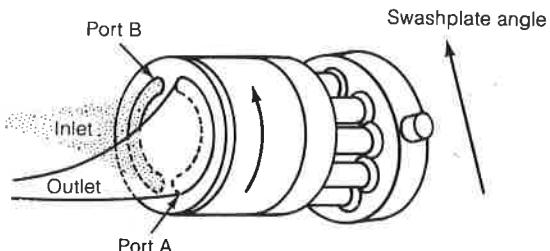
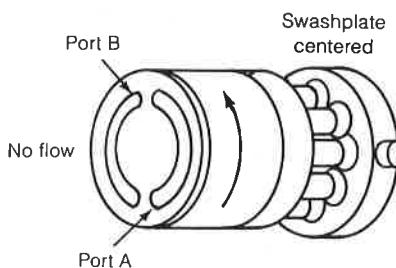
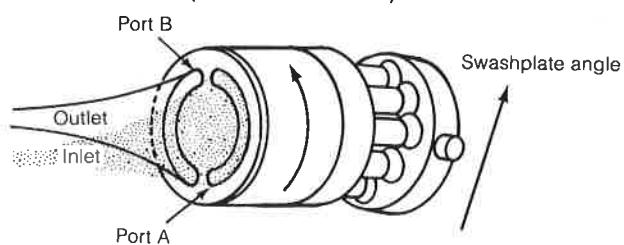


Figure 16.10.6 Exploded view of the elements of an axial piston pump (top) and assembled (bottom) to show operation of the pistons as the cylinder barrel rotates. Source: Parker-Hannifin Corp., Cleveland, OH.

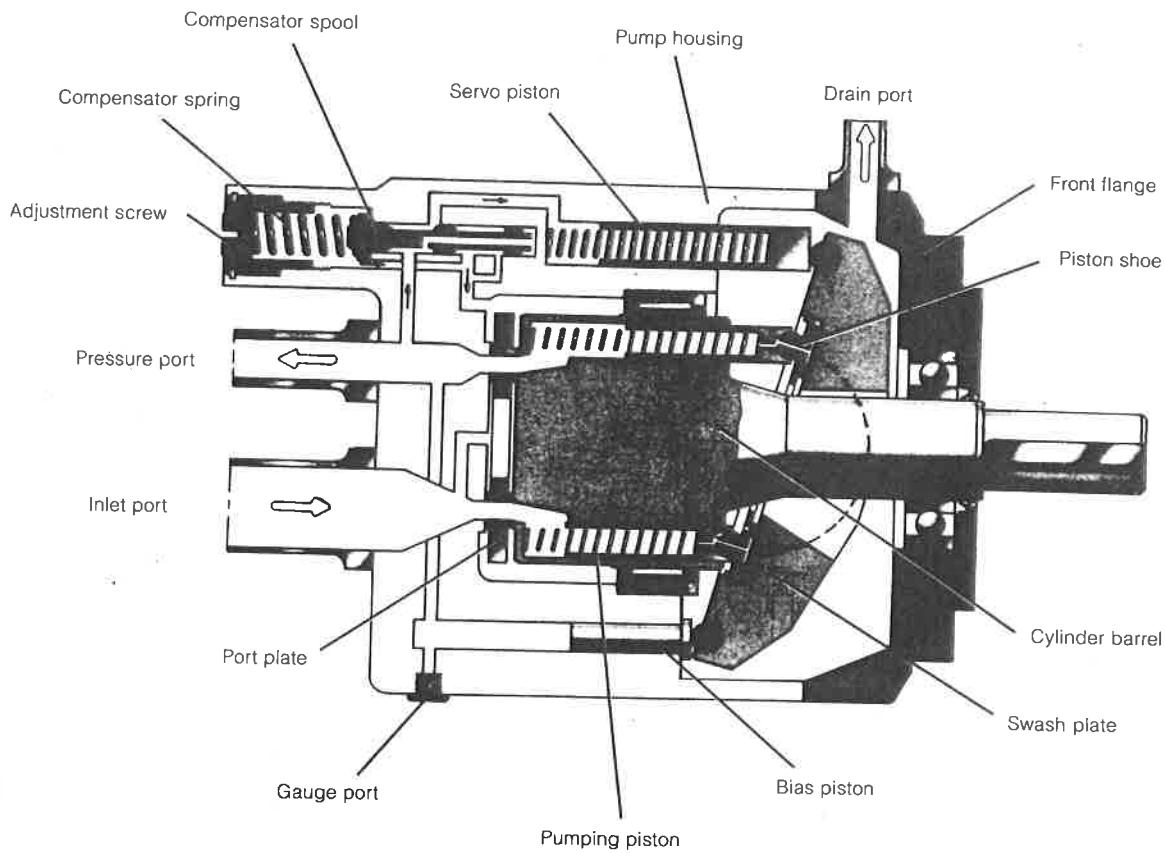


Figure 16.10.7 Cross-sectional drawing of an axial piston pump with compensation. The servo piston controls the angle of the swash plate. Source: Parker-Hannifin Corp., Cleveland, OH.

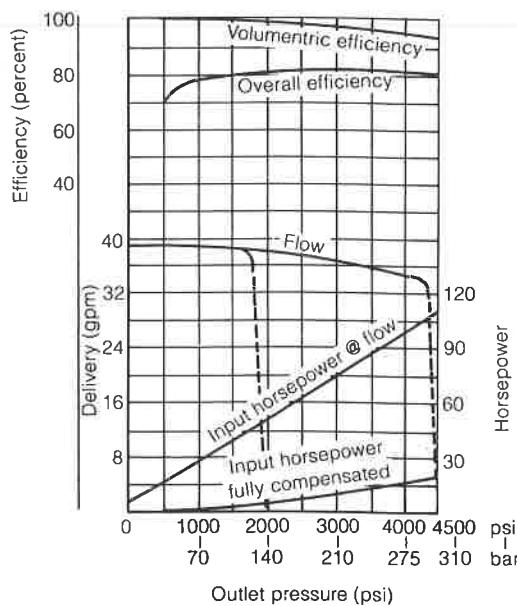


Figure 16.21.2 Performance curves for an axial piston, variable volume, compensated hydraulic pump. Source: Parker-Hannifin Corp., Cleveland, OH.