

## Lesson 3

The atmosphere

# The atmosphere on Earth

Dry Air (= Atmospheric air - H<sub>2</sub>O - contaminants (dust, pollen, ...))

	Mole Fraction	Molecular Weight $\frac{kg}{kmol} \approx \frac{lbm}{lbmol}$	
Nitrogen	78.08%	28.02	<chem>N#N</chem> strong triple bond!
Oxygen	20.95%	32.0	<chem>O=O</chem> double bond
Argon	0.93%	39.94	Greek αργον "inactive" <del>Ar</del> Ar no bond
Carbon Dioxide	0.03%	44.01	<chem>O=C=O</chem> linear shape double bond
Other	0.01%		
	<hr/> 100%		

$$\text{Apparent Molecular Weight} = \sum a_i M_i$$

$$M \approx 28.02 \cdot 0.7808 + 32.0 \cdot 0.2095 + 39.94 \cdot 0.0093 + 44.01 \cdot 0.0003$$

$$\approx 28.97 \frac{lbm}{lbmol} = 28.97 \frac{kg}{mol}$$

Gas Constant for air

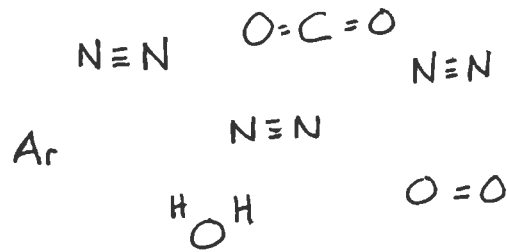
$$R = \frac{\bar{R}}{M} = \frac{1545.34 \frac{ft \cdot lbf}{R \cdot lbmol}}{28.97 \frac{lbm}{lbmol}} = \frac{1716.5 \frac{ft \cdot lbf}{R \cdot slug}}{28.97 \frac{lbm}{lbmol}} = 53.35 \frac{ft \cdot lbf}{R \cdot lbm}$$

Air density SSL (14.696 psi, 59°F)

$$\rho = \frac{PM}{\bar{R}T} = \frac{14.696 \text{ psi} \cdot 28.97 \text{ lbm}}{53.35 \frac{ft \cdot lbf}{R \cdot lbm} \cdot 518.67 \text{ R}} = 0.00237 \frac{slug}{ft^3}$$

# Wet air

The addition of water vapor changes the properties of "air":



Water vapor behaves as an ideal gas, thus we can model the mixture as an IG.

$$p = \frac{P_{\text{dry}}}{R_{\text{dry}} T_{\text{dry}}} + \frac{P_{\text{vapor}}}{R_{\text{vapor}} T_{\text{vapor}}}$$

where the partial pressures add to total pressure

$$P = P_{\text{dry}} + P_{\text{vapor}}$$

$$= \frac{P_{\text{dry}} M_{\text{dry}}}{R T_{\text{dry}}} + \frac{P_{\text{vapor}} M_{\text{vapor}}}{R T_{\text{vapor}}}$$

Temps are identical

$$= \frac{P_{\text{dry}} M_{\text{dry}} + P_{\text{vapor}} M_{\text{vapor}}}{\bar{R} T}$$

Partial pressure of vapor

$$P_{\text{vapor}} = \phi P_{\text{sat}}$$

$\phi \equiv$  relative humidity

$$= \frac{(P - P_{\text{vapor}}) M_{\text{dry}} + P_{\text{vapor}} M_{\text{vapor}}}{\bar{R} T}$$

$$= \frac{(P - \phi P_{\text{sat}}) M_{\text{dry}} + \phi P_{\text{sat}} M_{\text{vapor}}}{\bar{R} T}$$

$$= \frac{P M_{\text{dry}} + \phi P_{\text{sat}} (M_{\text{vapor}} - M_{\text{dry}})}{\bar{R} T}$$

$$M_{\text{dry}} = 28.97$$

$$M_{\text{vapor}} = 18.0$$

Thus,  $M_{\text{vapor}} - M_{\text{dry}}$  is negative

Increasing the water vapor decreases air density

Saturated Water Vapor  $\Rightarrow$  Partial Pressure

Arden - Buck Equation (curve fit)

$$P_s^{[kPa]}(T[^\circ C]) = 6.1121 \exp\left(\left(18.678 - \frac{T}{234.5}\right)\left(\frac{T}{257.14 + T}\right)\right)$$

$$P_s^{[psi]}(T[R]) = 0.08865 \exp\left(\frac{-0.002369(T - 8375.65)(T - 491.67)}{T - 28.818}\right)$$

Ex:

What is the partial pressure of water vapor at  $100^\circ F$ ?

$$P_s(100^\circ) = 0.08865 \exp\left(\frac{-0.002369 \left(\frac{559.67}{100} - 8375.65\right) \left(\frac{559.67}{100} - 491.67\right)}{\frac{100}{559.67} - 28.818}\right)$$
$$= 0.9502 \text{ psi}$$

$P_s(100^\circ)$  from my thermodynamics <sup>steam</sup> table is 0.9503 psi

Ex: What is  $P_{s_{H_2O}}$  at  $212^\circ F$ ? Hint: Boiling Water

You don't need the formula!  $P_s = P_{sL} = 14.7 \text{ psi} = 1 \text{ atm}$

Ex: At what altitude must you fly to boil water in your hand?

Human body  $\approx 98^\circ F$

$$P_s(98^\circ F) = 0.89 \text{ psi}$$

Consult a standard atmosphere table

$$h \approx 63000 \text{ ft}$$

please don't try at home!

## Wet air (continued)

$$\rho = \frac{P M_{\text{dry}} + \phi \cdot 0.08865 \exp\left(\frac{-0.002369(T - 8375.65)(T - 491.67)}{T - 28.818}\right) (M_{\text{vapor}} - M_{\text{dry}})}{\bar{R}T}$$

function of  $\rho, \phi, T$

Ex:  $\rho = 14.696 \text{ psi}, 90^\circ\text{F}, 90\% \text{ rh}$

$$\rho = \dots = 0.00220 \frac{\text{slug}}{\text{ft}^3}$$

Ex: " " 0% rh

$$\rho = 0.002243 \frac{\text{slug}}{\text{ft}^3}$$

# Impact of Wet Air.

	Temp	rh	Alt	$\rho$	% SSL
• Standard Sea Level (SSL) Std-Day	59°F	0%	0 ft <sub>MSL</sub>	$\rho = 0.00237 \frac{\text{slug}}{\text{ft}^3}$	100
• Alabama Summer (hot + humid)	90°F	90%	$\approx 0$ ft <sub>MSL</sub>	$\rho = 0.002206 \frac{\text{slug}}{\text{ft}^3}$	93%
• " " Dry	90°F	0%		$\rho = 0.00224 \frac{\text{slug}}{\text{ft}^3}$	94%
• Alabama Winter (Wet)	40°F	90%		$\rho = 0.00246 \frac{\text{slug}}{\text{ft}^3}$	104%
• " " Dry	40°F	0%		$\rho = 0.00246 \frac{\text{slug}}{\text{ft}^3}$	104%
• Antarctica (cold + dry)	-126°F	0%		$\rho = 0.00369 \frac{\text{slug}}{\text{ft}^3}$	155%
• Denver, CO (Std day)	40°F	0%	5000 ft <sub>MSL</sub>	$\rho \approx 0.00205 \frac{\text{slug}}{\text{ft}^3}$	86%

## ASHRAE Chart Comparison



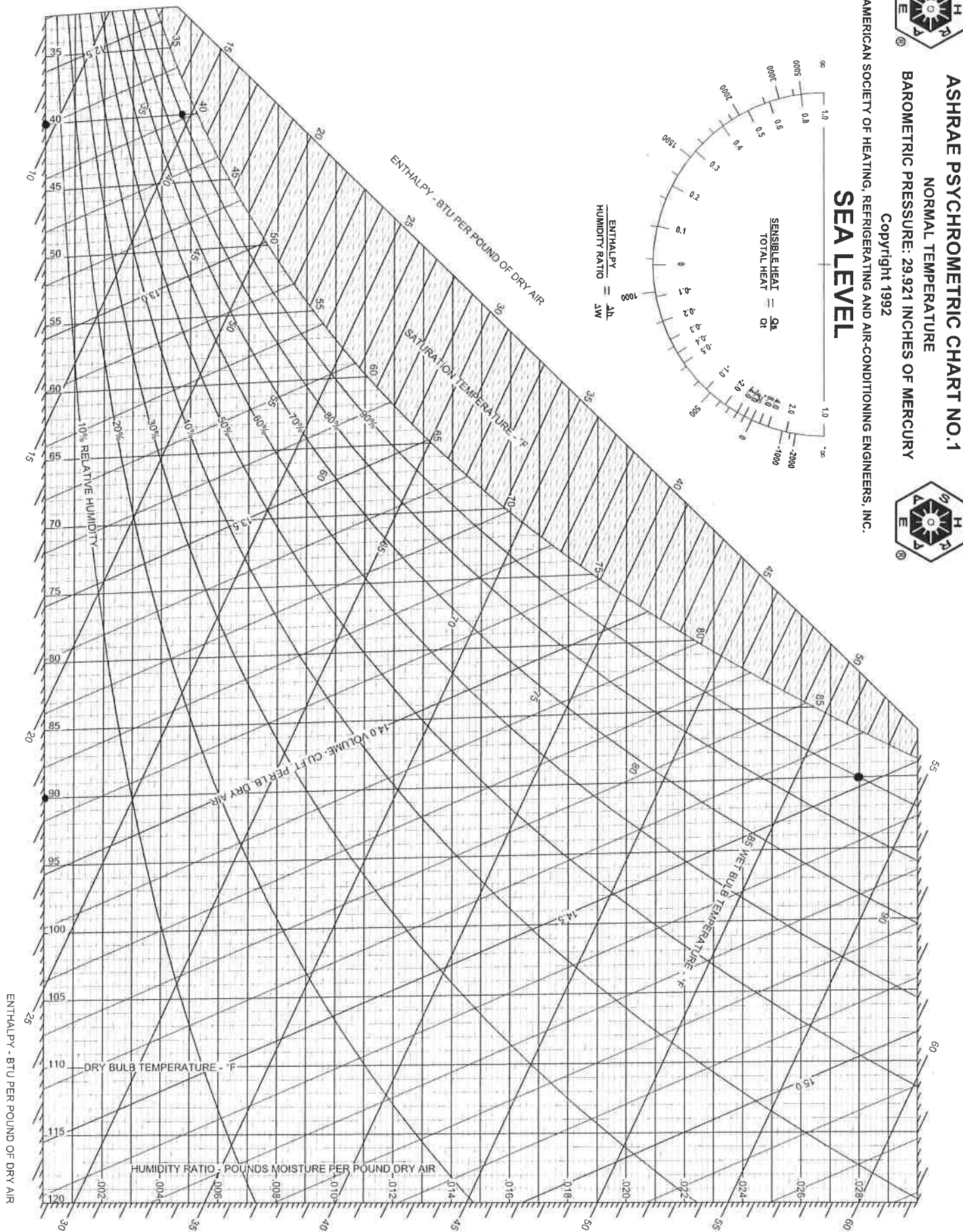
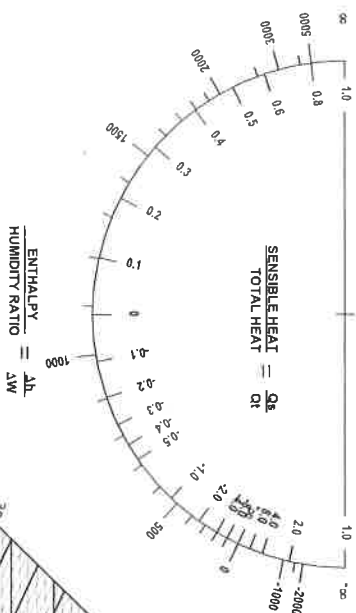
# ASHRAE PSYCHROMETRIC CHART NO. 1

NORMAL TEMPERATURE  
BAROMETRIC PRESSURE: 29.921 INCHES OF MERCURY

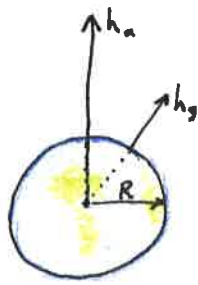
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## SEA LEVEL



# Atmosphere

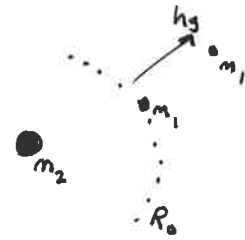


Distinguish between absolute altitude from center of Earth ( $h_a$ ) and altitude from the Earth's surface ( $h_g$ )

$$h_a = h_g + R$$

Gravity is not constant wrt altitude.

$$F = G \frac{m_1 m_2}{r^2}$$



How does gravity change with altitude?

$$F_{R_0} = \frac{G m_1 m_2}{R_0^2} \quad \text{force at } R_0$$

$$F_{R_0+h_g} = \frac{G m_1 m_2}{(R_0+h_g)^2} \quad \text{force at } R_0+h_g$$

$$\Rightarrow \frac{G m_2}{R_0^2} = g_0$$

$$g = \frac{G m_2}{(R_0+h_g)^2} = g_0 \frac{R_0^2}{(R_0+h_g)^2}$$

$$g = g_0 \frac{R_0^2}{(R_0+h_g)^2}$$

Define a new altitude called "geopotential altitude",  $h$ , where gravity is constant ( $g_0$ ).

$$\underset{\substack{\uparrow \\ \text{geopotential}}}{dh} = \frac{g}{g_0} \underset{\substack{\uparrow \\ \text{geometric}}}{dh_g} \Rightarrow dh = \frac{R_0^2 dh_g}{(R_0+h_g)^2} \Rightarrow h = \frac{R_0}{R_0+h_g} h_g$$

Negligible for most aircraft applications.

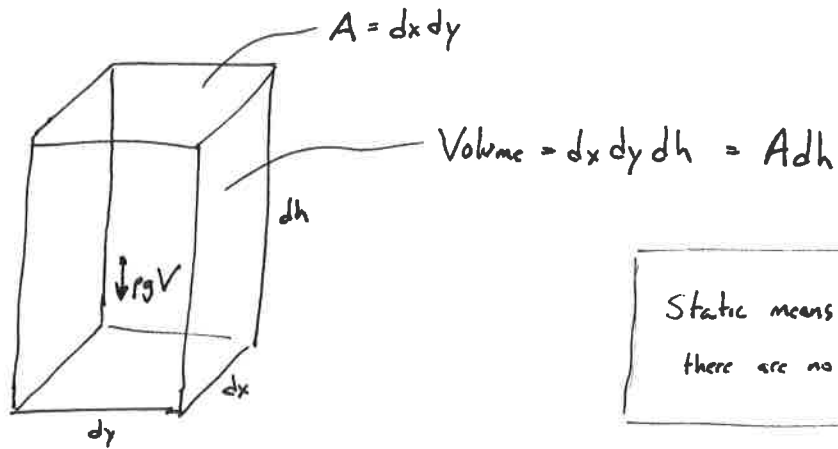
$\approx 0.1\%$  error at 21000 ft

1% error at 200000 ft

Nevertheless, we will use  $h$  (geopotential) altitude.



# Static Column of Fluid



Static means no velocity, so no  $\frac{du_i}{dx_j}$ , thus there are no viscous forces.

Summation of forces in  $h$  direction

$$P_{\text{bottom}} A = P_{\text{top}} A + \rho g V$$

Taylor Series expansion for  $P_{\text{top}} = P_{\text{bottom}} + \frac{dP}{dh} dh$

$$PA = \left( P + \frac{dP}{dh} dh \right) A + \rho g A dh$$

Reduce (divide by  $A$ , cancel  $PA$  terms)

$$\frac{dP}{dh} dh + \rho g dh = 0$$

Divide by  $dh$

$$\frac{dP}{dh} + \rho g = 0$$

Gov Egu

$$dp = -\rho g dh$$

# Atmosphere (continued)

$$dp = -\rho g_0 dh$$

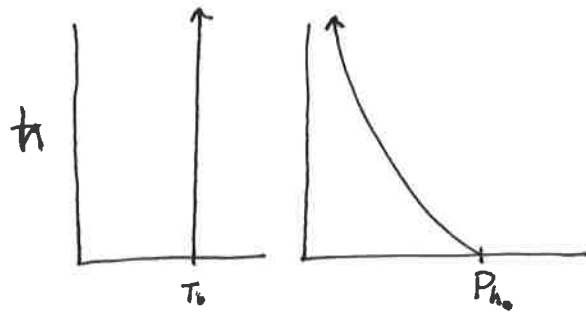
Ideal Gas is  $p = \rho RT$ , substitute for  $\rho$

$$dp = -\frac{p}{RT} g_0 dh \quad \Rightarrow \quad \frac{dp}{p} = -\frac{g_0 dh}{RT}$$

- Isothermal  
Zero lapse rate ( $\Delta T \neq f(h)$ ) ( $T = T_0$ )

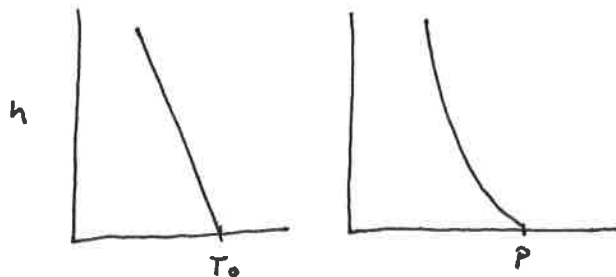
$$\frac{dp}{p} = \frac{-g_0}{RT_0} dh \quad \text{Integrate} \quad \ln \frac{p}{p_0} = \left. -\frac{g_0}{RT_0} h \right|_{h_0}^{h_1}$$

$$\underbrace{\ln p_1 - \ln p_0}_{\ln \frac{p_1}{p_0}} = \frac{-g_0}{RT_0} (h_1 - h_0) \quad \Rightarrow \quad p_{h_1} = p_{h_0} e^{-\frac{g_0 (h_1 - h_0)}{RT_0}}$$



- Linear lapse rate ( $T = T_0 + \lambda(h_1 - h_0)$ )

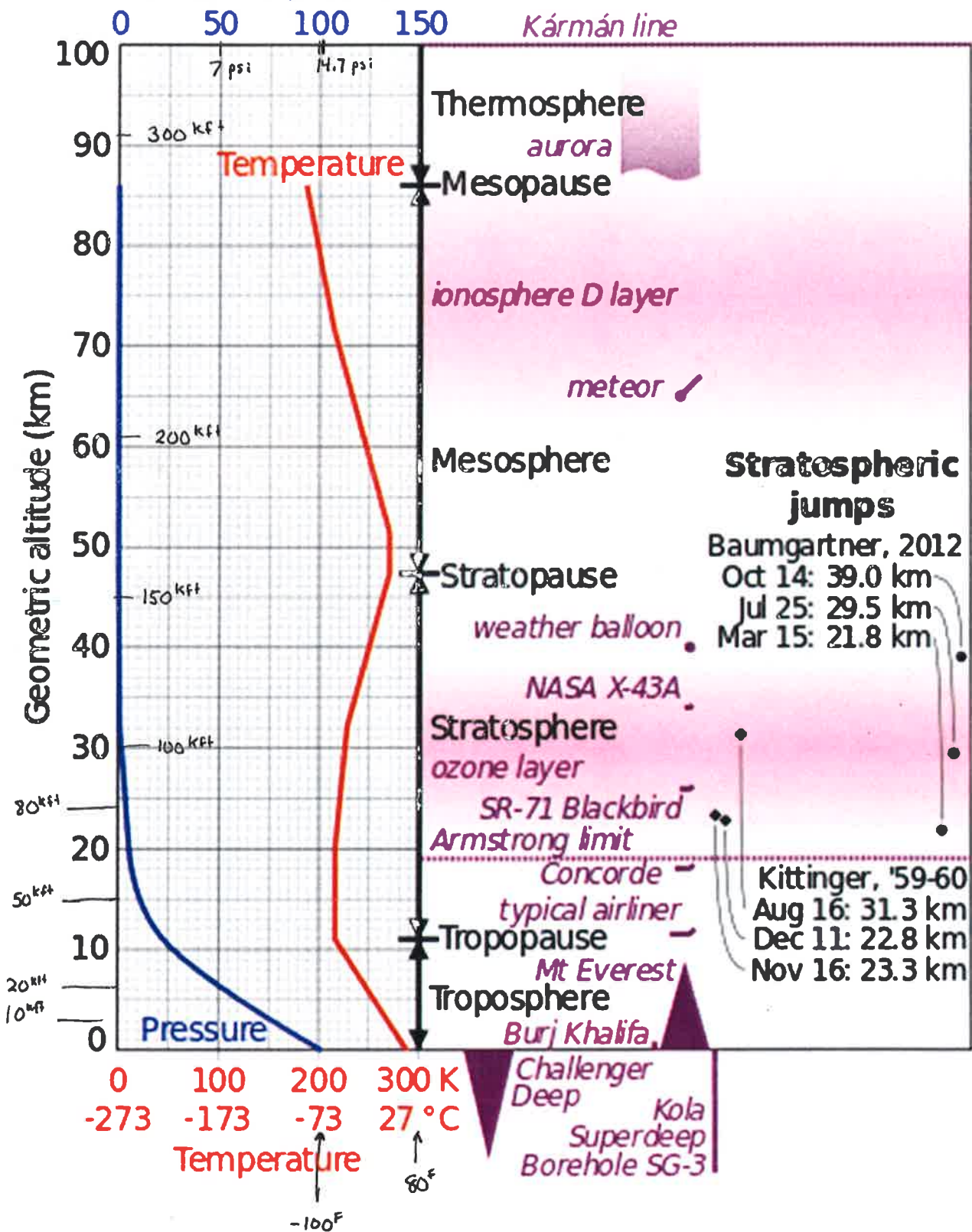
$$\frac{dp}{p} = \frac{-g_0}{R(T_0 + \lambda(h - h_0))} dh \quad \text{Integrate (slightly involved)} \quad \frac{p}{p_0} = \left( \frac{T}{T_0} \right)^{-\frac{g_0}{R\lambda}}$$



# International Standard Atmosphere

SpaceShipOne •

Pressure (kN/m<sup>2</sup>)

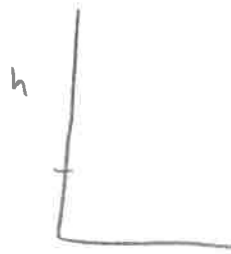


## Standard Atmosphere

$h$ ft	$\lambda$ R/ft
0	-0.003566
36089	0
65617	+0.000549
104987	+0.001586

$$P(0\text{ft}) = 14.696 \text{ psi}$$

$$T(0\text{ft}) = 59^\circ\text{F}$$



For more information, refer to the 1976 standard atmosphere.

## Reality:

The atmosphere is not, has not, and will never be "standard" or static.

metogram.

For engineering purposes, additional standard atmosphere models exist:

Hot Day

Cold Day

Tropics

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