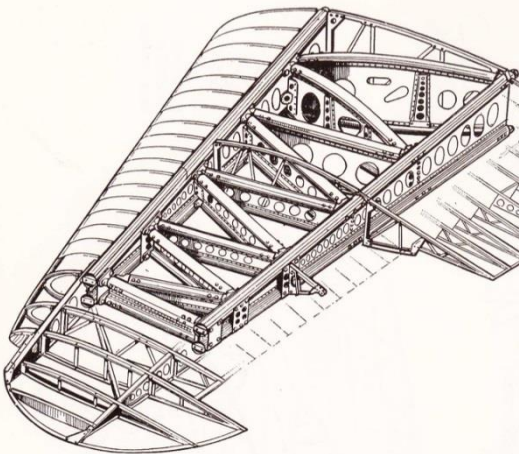


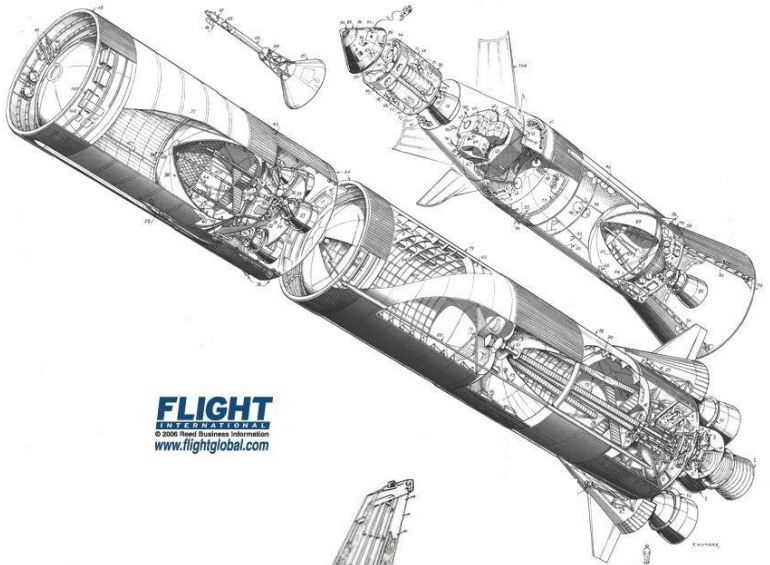
# Aerospace Structures

## AEM 341

### Spring 2018



*Understanding Aircraft Structures, John Cutler*



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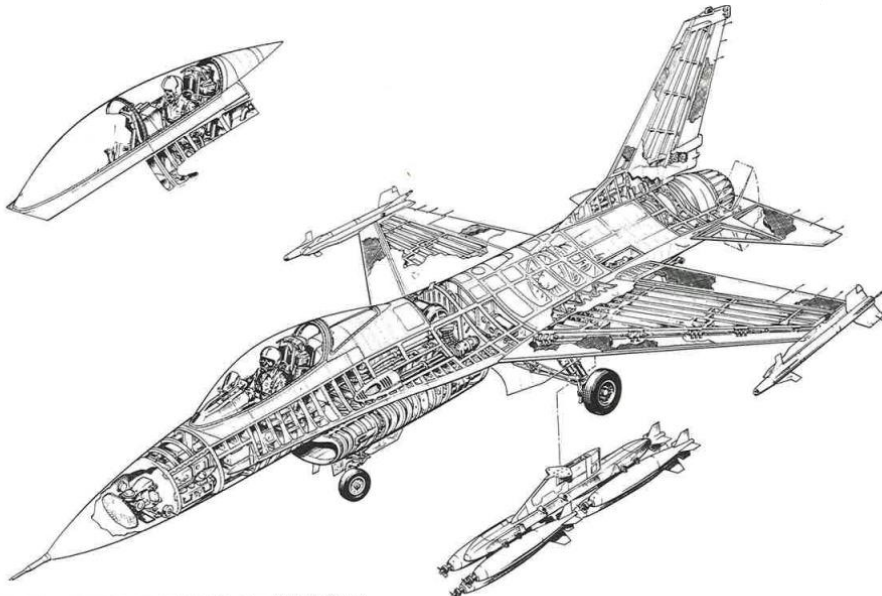


FIG. 1.1. General Dynamics F-16 (Courtesy U.S. Air Force).



## Barn Owl

- Perch
- Launch
- Flapping Flight
- Catch & Eat
- Stealth (night + sound)

Warren Photographic

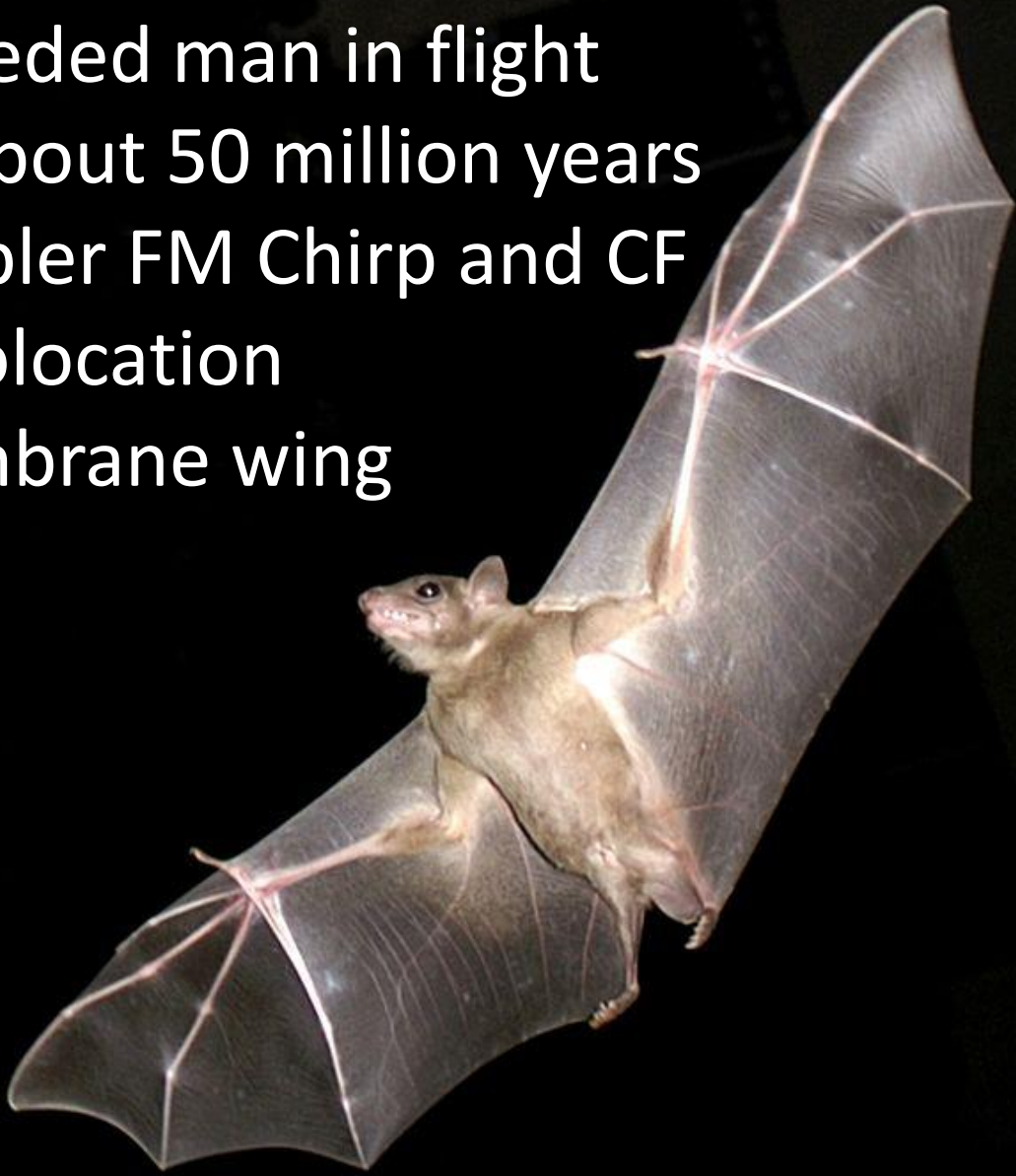


## Falcon

- +242 mph dive
- +Faster than man until 1923

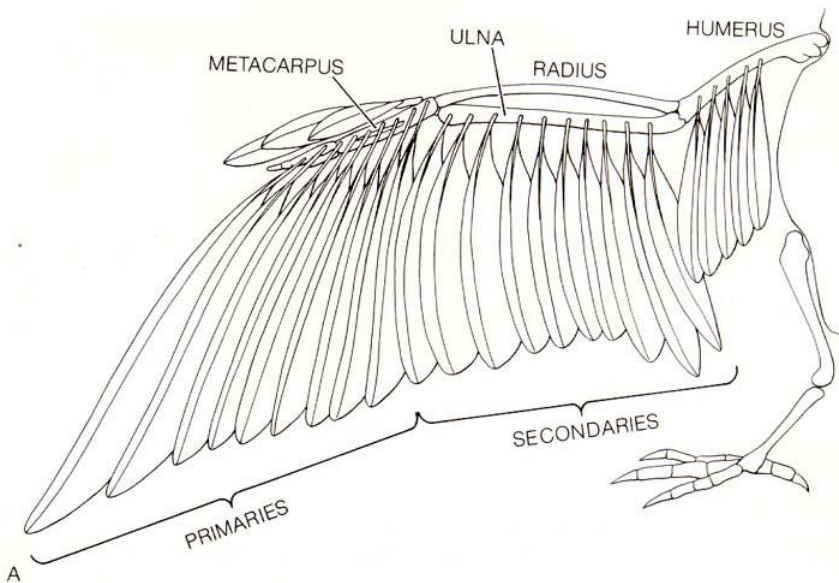
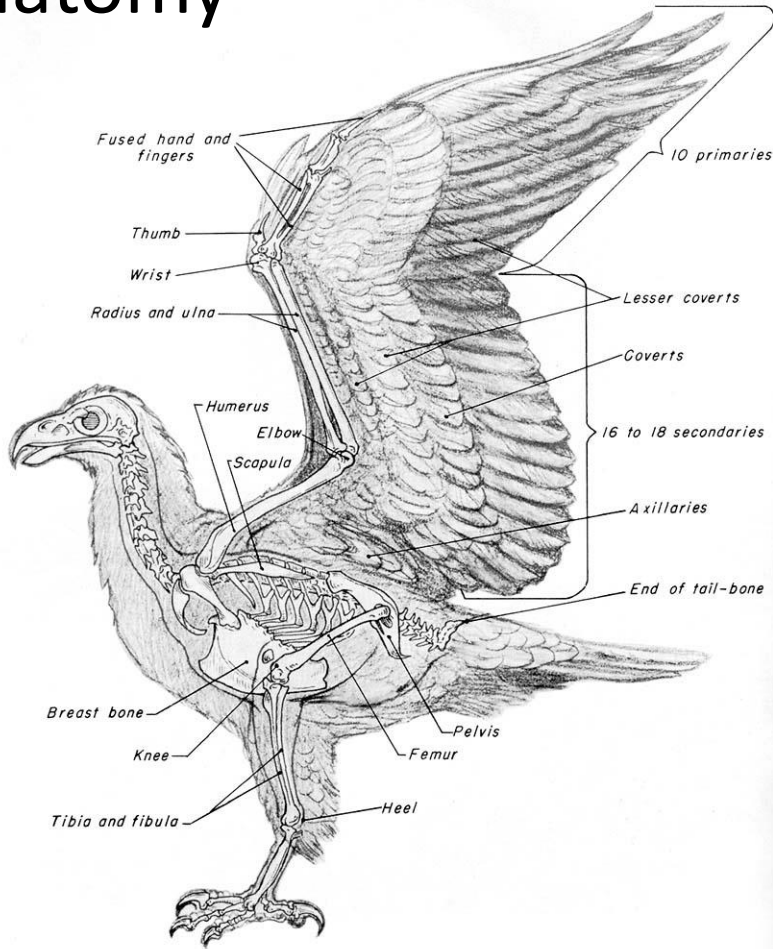
# Bat

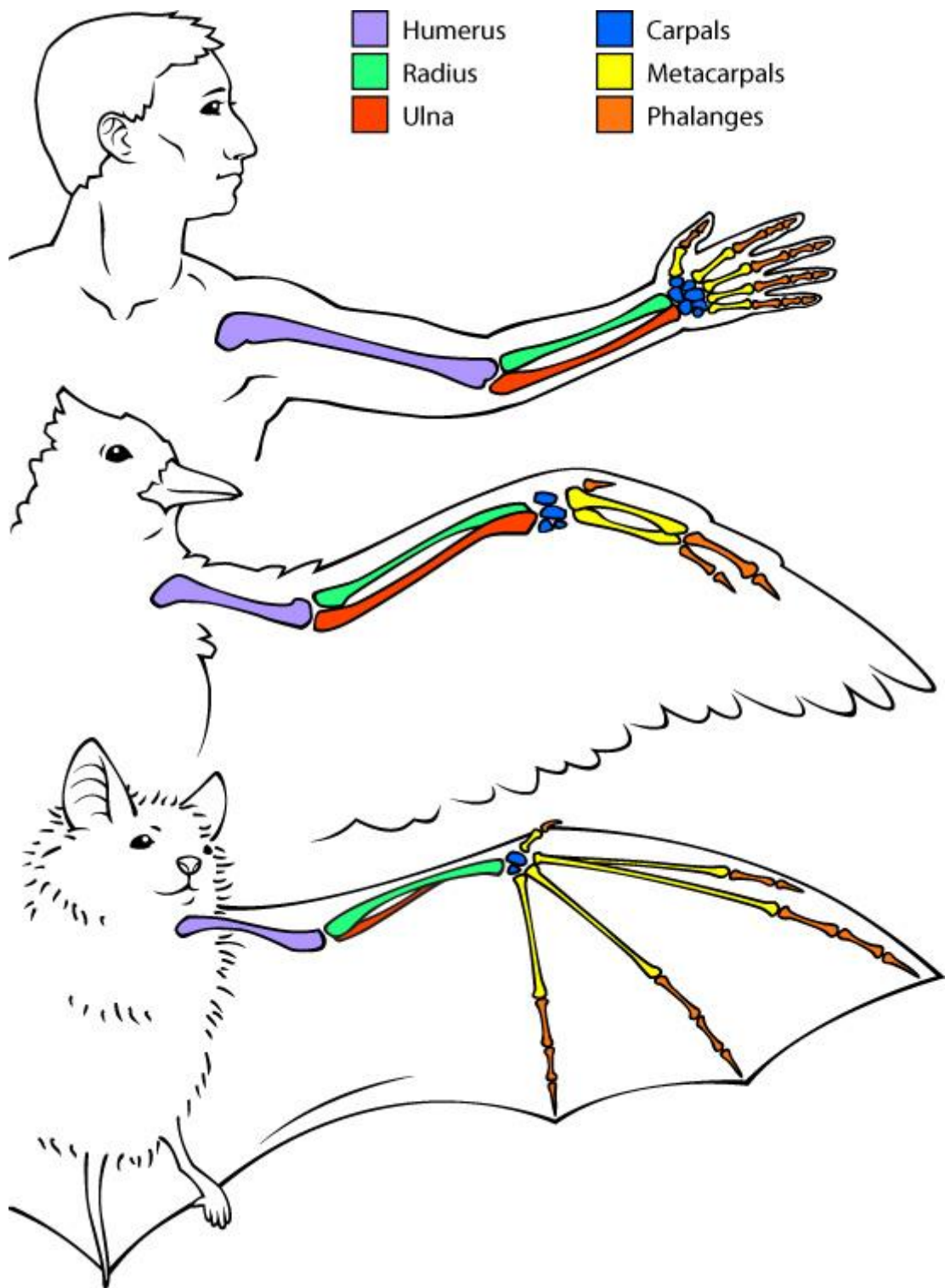
- Preceded man in flight by about 50 million years
- Doppler FM Chirp and CF echolocation
- Membrane wing



By Original photo: אורן פלס Oren Peles Derivative work: User:MathKnight - File:PikiWiki Israel 11327 Wildlife and Plants of Israel.JPG by אורן פלס, CC BY 2.5, <https://commons.wikimedia.org/w/index.php?curid=27775788>

# Bird Anatomy



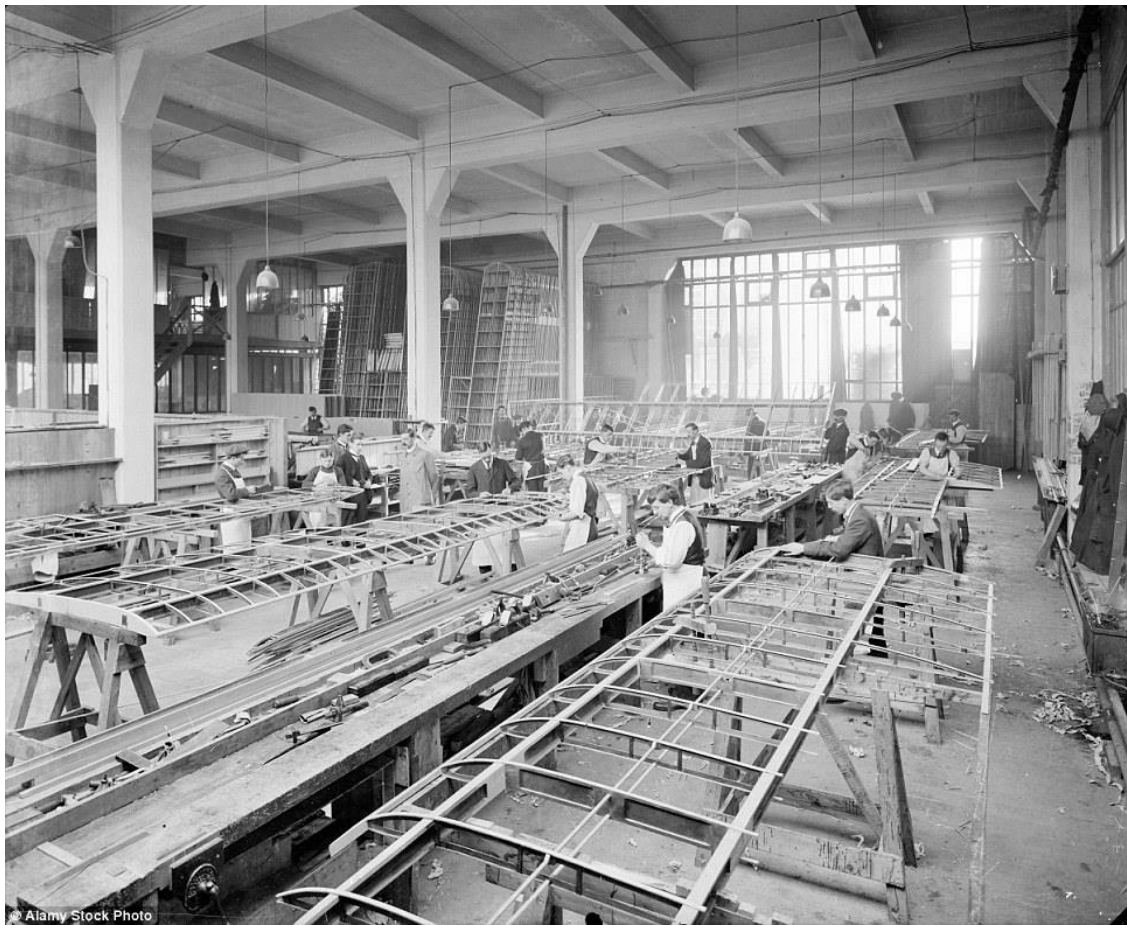


# Human Flight

## Wright Flyer 1903



# WW1 Technology



# Fokker Dr I (June 1917)

## Germany

Seen in its fighter context as a formula for combining good climbing qualities with extreme lateral manoeuvrability made possible by an exceedingly small overall wing span, the triplane was an aberration from the mainstream of fighter development. It was to enjoy a brief heyday in 1917 and be obsolescent before that year's end. Indeed, of many fighter triplanes developed, but two were to see combat.

Triplane investigation in Germany dated from aviation's early pioneering days, but the catalyst in its further development for the fighter role was provided by the February 1917 operational debut of the Royal Naval Air Service's Sopwith Triplane. The *Fliegertruppen* were startled by the remarkable manoeuvrability and climb rate demonstrated by the Sopwith. Germany being panicked into launching a massive single-seat fighter triplane development effort, which, with the sole exception of one type, the Fokker Dr I, was to prove an abysmal failure. The Dr I was thus conceptually unoriginal in being engendered by the Sopwith, but it nevertheless embodied some highly innovative features.

Designed by Reinhold Platz, the prototype—at that time

known as the D VI, Fokker having skill to adopt V-series designations for experimental aircraft—was an outstandingly compact triplane with verespansions (literally "without bracing") or cantilever wings dispensing with flying landing and incidence wires, the necessary strength being imparted by an original single-spar arrangement, which was actually two boxspars joined vertically. The fuselage was of welded steel tubing with transverse bracing to form a rigid box-spider structure. Wing oscillation in certain flight regimes dictated introduction of thin, non-structural spruce I-type interplane struts to provide the desired rigidity.

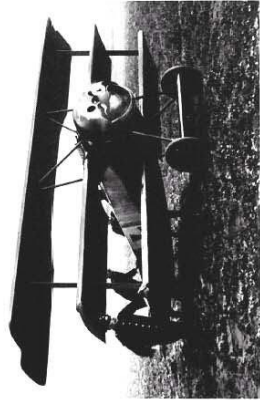
Enjoying the patronage of no less a personality than Manfred Freiherr von Richthofen, the triplane was ordered into production on 14 July 1917, two prototypes being tested at the Front in the following month by von Richthofen and Werner Voss. Production Dr I was reached the Front in October proving very sensitive about all axes and most taxing to fly. The Dr I, nevertheless, possessed superlative aerobatic qualities, and if a slow, low-altitude performer, it

made a dangerous adversary with a skilled pilot at its controls. Its manoeuvrability was arguably second to no other fighter. In developing the Dr I, however, the *Fliegertruppe* had not comprehended the inherent limitations of the triplane configuration and its early demise was inevitable, only 320 fighters of this type being built.

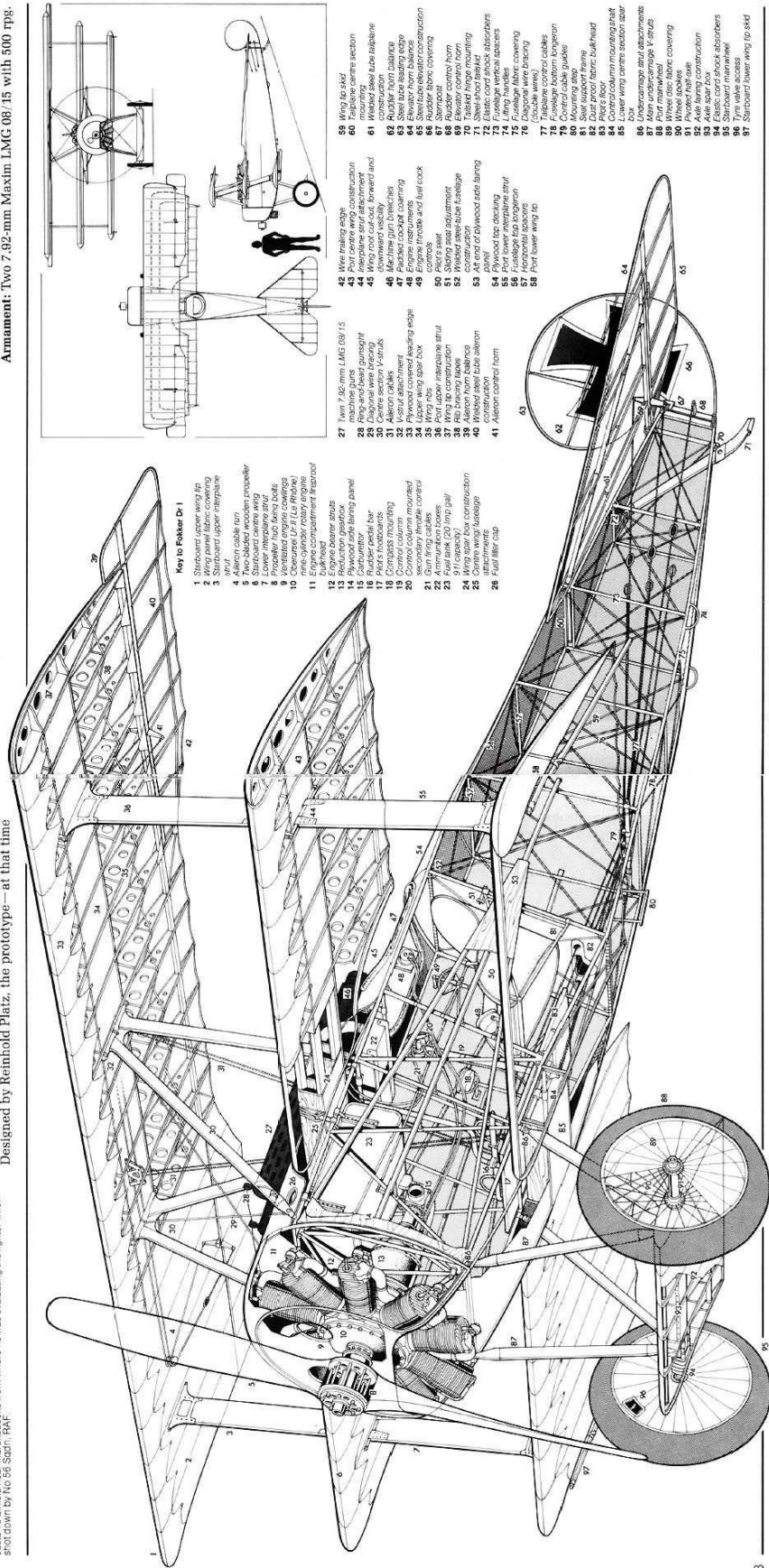
### SPECIFICATION: Fokker Dr I

**Power Plant:** One Oberursel Ur II nine-cylinder rotary air-cooled engine rated at 110 hp at 1,200 rpm for take-off. Two-bladed fixed-pitch wooden propeller. Internal fuel capacity, 20 Imp gal (91 l).

**Performance:** Max speed, 115 mph (185 km/h) at sea level, 102.5 mph (165 km/h) at 13,125 ft (4,000 m), initial climb, 1,800 ft/min (9.15 m/sec); time to 3,280 ft (1,000 m), 2.9 min, to 9,840 ft (3,000 m), 10.1 min, to 16,405 ft (5,000 m), 23.85 min; range, 185 mls (300 km).  
**Weights:** Empty, 894 lb (406 kg); loaded, 1,291 lb (586 kg).  
**Dimensions:** Span, 23 ft 7 in (7.19 m); length, 18 ft 11 in (5.77 m); height, 9 ft 6 in (2.95 m); wing area, 200.86 sq ft.  
**Armament:** Two 7.92-mm Maxim LMG 08/15 with 500 rpg.



Above: One of the Dr I prototypes (F 1103/17) being taxied by Werner Voss at the Front. Below: The engine from where he was evaluating the fighter when shot down by No 55 Sqn, RAF.



#### Key to Fokker Dr I

- 1 Standard upper wing to machine guns
- 2 Wing panel fabric covering
- 3 Standard upper interplane struts
- 4 Aileron cable run
- 5 Two-bladed wooden propeller
- 6 Lower interplane struts
- 7 Lower interplane struts
- 8 Propeller hub
- 9 Propeller hub
- 10 Oberursel Ur II engine
- 11 Engine-propeller shaft
- 12 Bulkhead
- 13 Reduction gearbox
- 14 Plywood side fairing panel
- 15 Plywood fuselage
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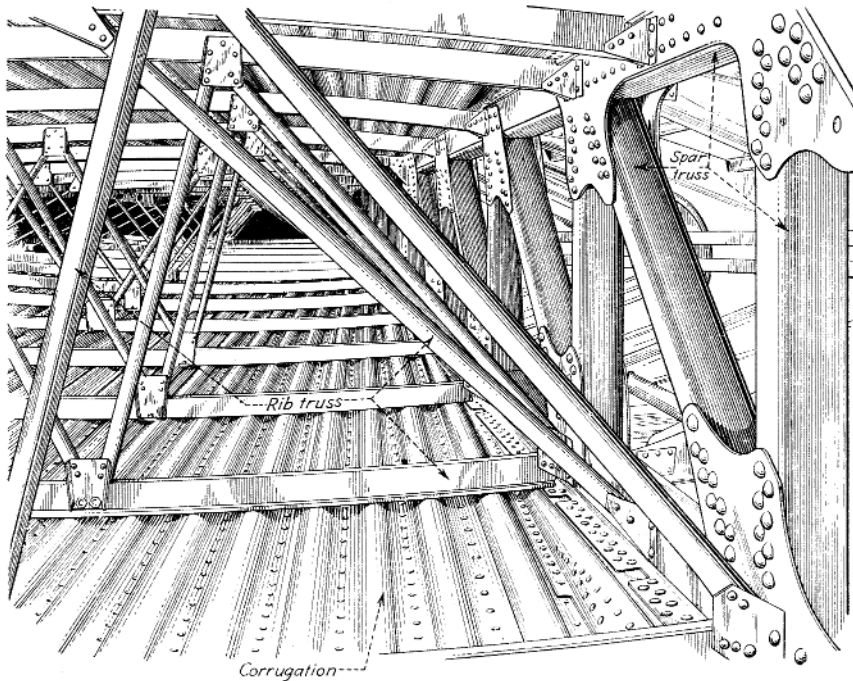


# Junkers J1 (1915)

Blechesel = Tin Donkey

+Metal Cantilever Wings

+2 decades advanced



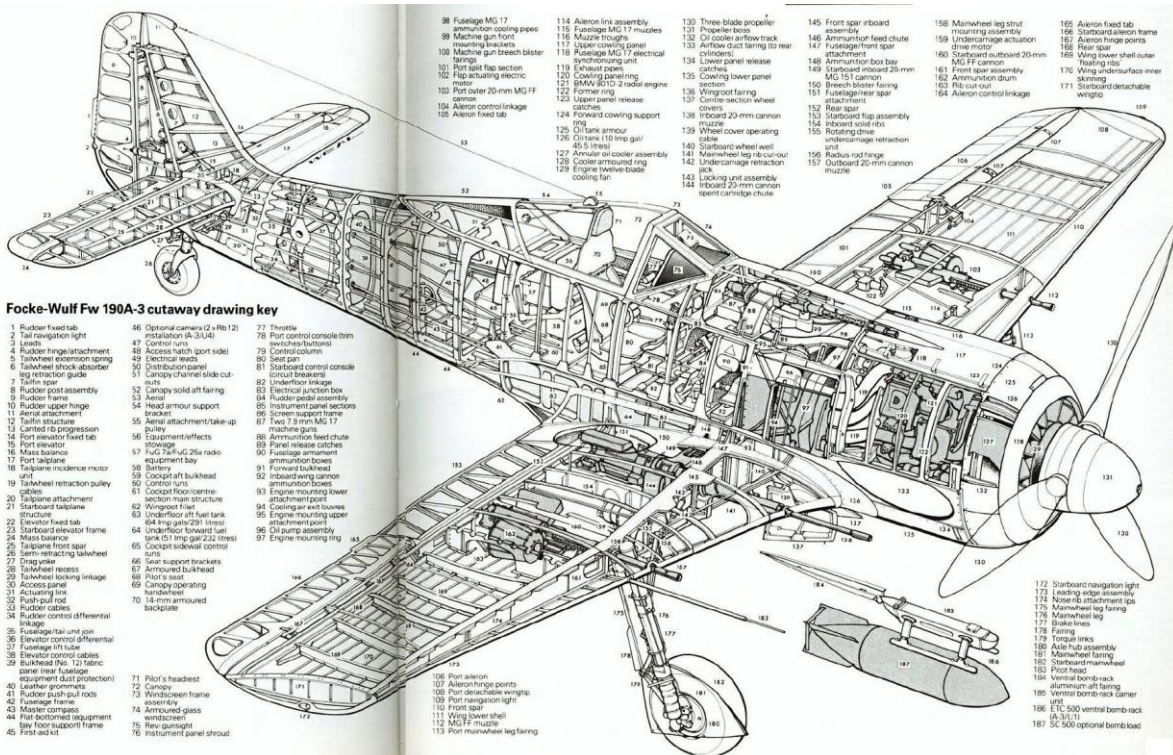
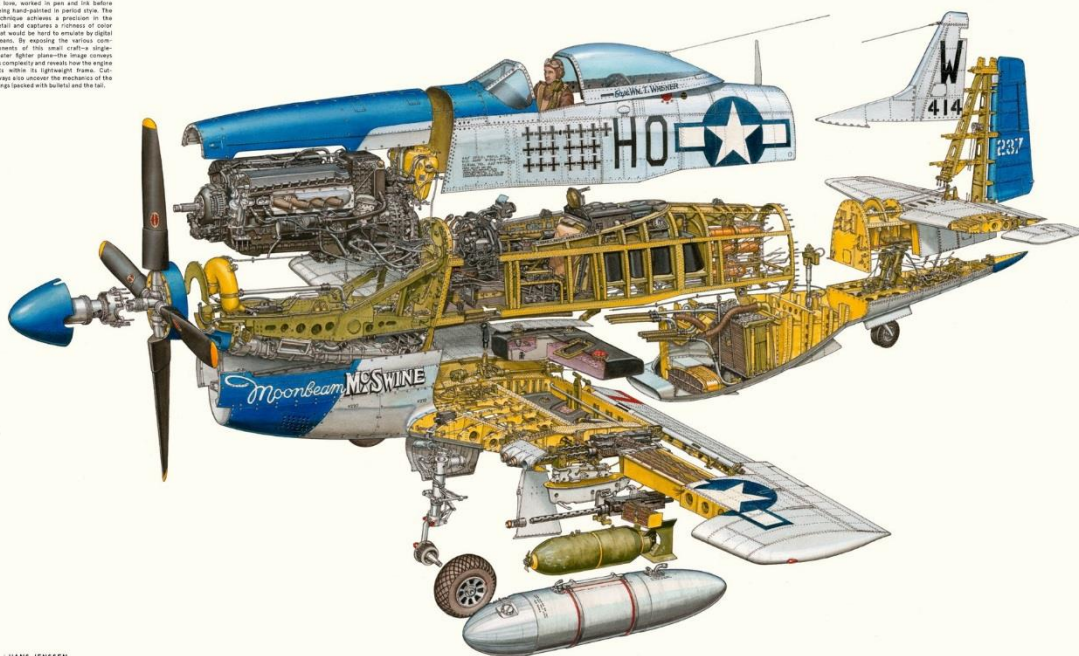
*Closeup of internal wing structure showing attachment details of spar truss, rib truss, and corrugations.*

# Mid 1940s Technology

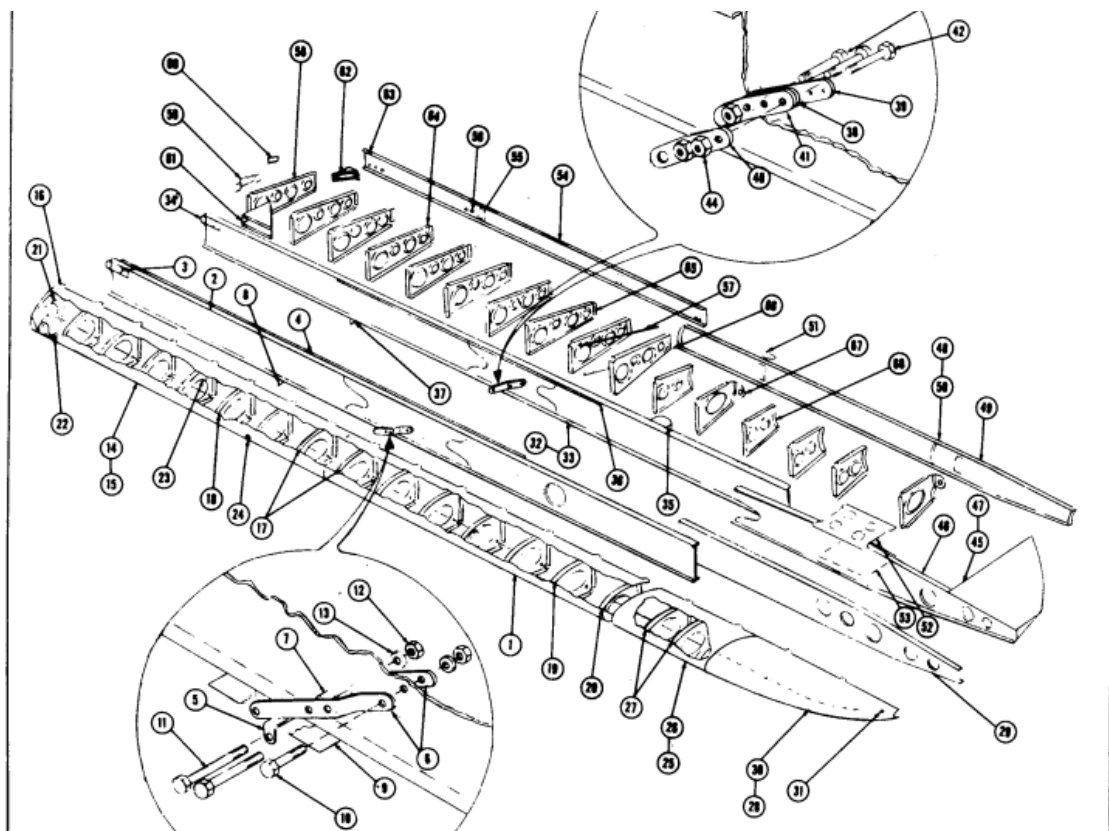
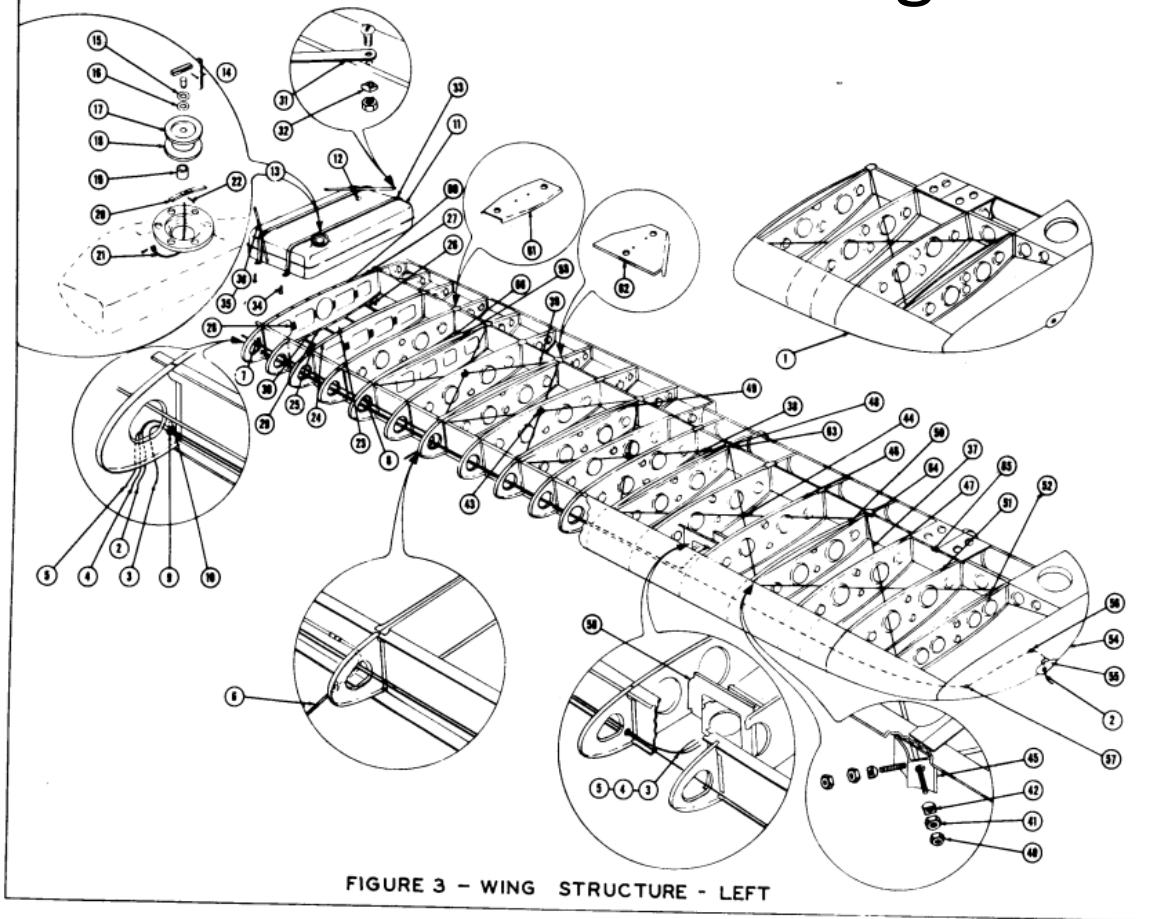
Here, Anson's exploded diagram of a twin-D30 Mustang fighter plane is a labor of love, worked in pen and ink before being transferred to plastic type. The technique achieves a precision in the detail and captures a richness of color that would be hard to emulate by digital means. By exposing the various components of this great craft—single-seater fighter airplane—Anson conveys its complexity and reveals how the engine sits within its airplane frame. Cut-aways also uncover the mechanics of the world-famous airfield and the tail.

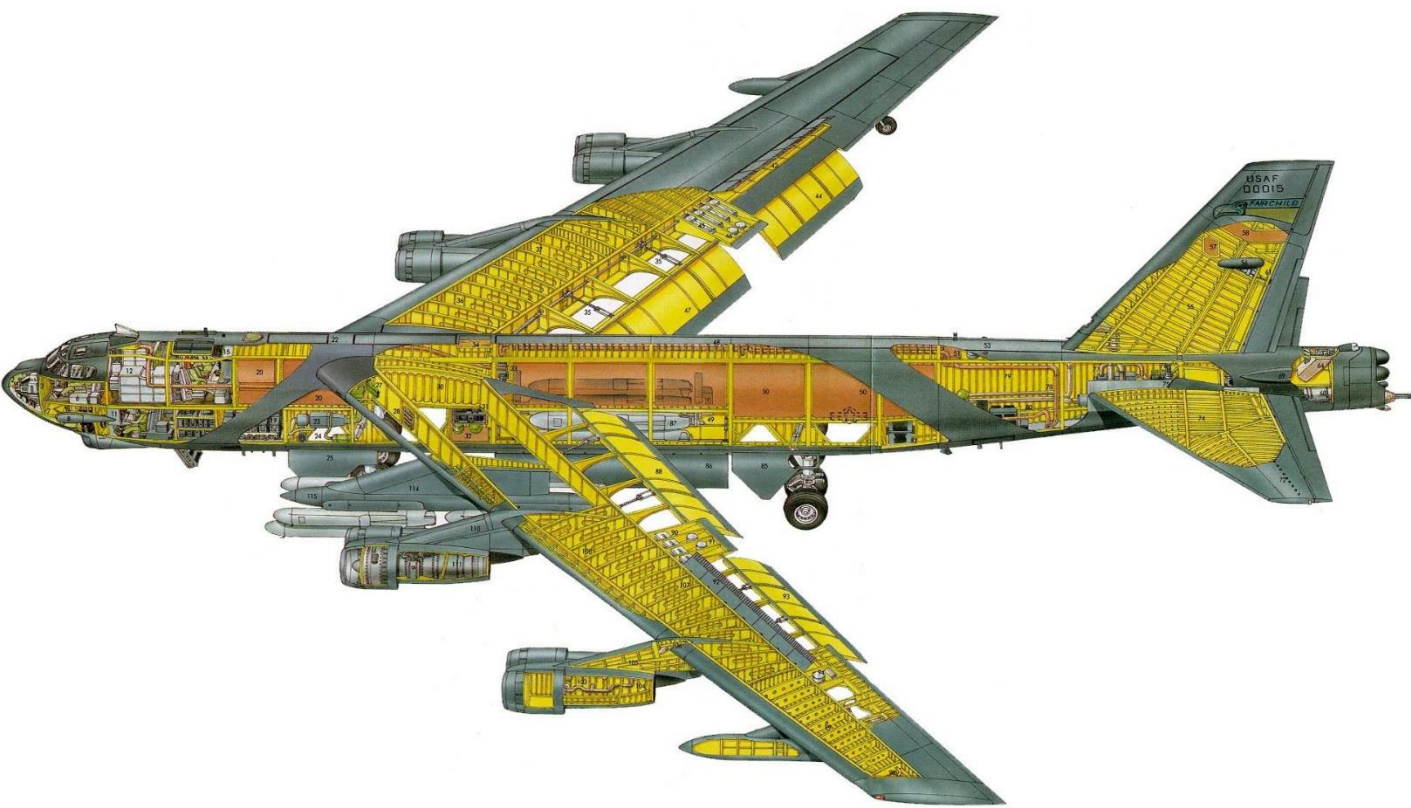
HANS JENSEN  
P-51 Mustang

49



# Cessna 170 Wing Structure





**Boeing KC-135 Cutaway Drawing Key**

- 1 Detachable radome
- 2 Weather radar scanner
- 3 Radar array
- 4 Forward pressure bulkhead
- 5 Forward fuselage structure
- 6 Pilot's side console
- 7 Rudder pedal assembly
- 8 Central console
- 9 Instrument panel
- 10 Windscreen defrost/rain removal
- 11 Windscreen panels
- 12 Co-pilot's seat
- 13 Eyebrow windows
- 14 Ditching external handholds
- 15 Overhead console
- 16 Crew instructor's seat
- 17 Pilot's seat
- 18 Crew instructor's console
- 19 Escape spools
- 20 Entry and escape hatch
- 21 Nosewheel hydraulic actuator
- 22 Entry/escape tunnel
- 23 Crew entry door
- 24 Ladder
- 25 Twin nosewheels
- 26 Fuselage frame
- 27 AC power shield
- 28 Electrical equipment racks
  
- 29 Supernumerary crew station
- 30 Navigator's stool (celestial observation)
- 31 Navigator's seat
- 32 Navigator's instrument panel
- 33 Overhead storage rack
- 34 Celestial observation windows
- 35 Window defrost
- 36 Electronics rack
- 37 Flight deck door
- 38 Circuit breaker panel
- 39 Toilet
- 40 Battery
- 41 Cargo door frame
- 42 Forward fuselage underfloor fuel tanks (four tanks) capacity 5,800 US gal (8 612 l)
- 43 Floor support beam
- 44 Fuel filler access 5,000 lb and 10,000 lb (2 268-4 540 kg)
- 45 Fuel filler access (fuselage illumination)
- 54 Oil filler access
- 55 Surge bleed outlet
- 54 Nacelle pylon
  
- 55 Pylon strut access
- 56 Wing leading-edge
- 57 No 2 main fuel tank capacity 2,275 US gal (8 612 l)
- 58 Fuel filler access
- 59 Starboard inner pylon light (fuselage illumination)
- 60 Pylon strut attachment
- 61 Wing front spar
- 62 No 4 main tank capacity 2,062 US gal (7 805 l)
- 63 Leading-edge wing flap
- 64 No 4 engine nacelle
- 65 No 4 engine fuel air starter fan
- 66 Oil cooler chin intake scoop
- 67 Engine inlet nose cowl
- 68 Nacelle pylon
- 69 Dry bays
- 70 No 4 reserve tank capacity 434 US gal (1 643 l)
- 71 Wing outer leading-edge
- 72 Starboard navigation light
- 73 Starboard wingtip
- 74 Starboard outer aileron
- 75 Aileron lockout mechanism
- 76 Flap torque tube linkage
- 77 Aileron balance tab
- 78 Actuating links
- 79 Bus quadrant
- 101 Keel beam antenna bay
- 102 Bulkhead section
- 103 Fuselage light (wing illumination)
- 104 Wingroot/fuselage fillet plate
- 105 Front spar pick-up point
- 106 Port landing lights
- 107 Air-conditioning pack (port) intake scoop
- 108 Leading-edge structure
- 109 Wing front spar
- 110 Inboard wing stringers
- 111 No 2 main tank inner wall capacity 2,275 US gal (8 612 l)
- 112 Stiffeners
- 113 Engine bleed air manifold
- 114 Water injection manifold
- 115 Four-wheel main landing-gear
- 116 Torson links
- 117 Landing-gear trunnion
- 118 Wing rear spar
- 119 Angled ribs
- 133 Wing ribs
- 134 Lower surface inner stringers
- 135 No 1 main tank capacity 2,062 US gal (7 805 l)
- 136 Front spar
- 137 Fuel filler access
- 138 Strengtheners
- 139 Dry bays
- 140 Pylon fairing
- 141 Strut access
- 142 Hinged cowling panels
- 143 Engine inlet nose cowl
- 144 Nacelle frame
- 145 Pylon structure
- 146 Pylon support strut
- 147 Fuel panels
- 148 Nacelle support aft fairing
- 149 Wing outer leading-edge
- 150 No 1 reserve tank capacity 434 US gal (1 643 l)
- 151 Rear spar
- 166 Port landing-gear well
- 167 Landing-gear pivot
- 168 Side strut
- 69 Undercarriage retraction strut
- 70 Rear spar pick-up point

- 152 Port navigation light
- 153 Port wingtip
- 154 Port outer aileron
- 155 Aileron control tab
- 156 Bus quadrant
- 157 Port outer double-slotted flap
- 158 Port outer spoilers
- 159 Port inner aileron
- 160 Aileron control tab
- 161 Hinge fairing
- 162 Port inner double-slotted flap
- 163 Port inner spoilers
- 164 Wing fillet flap
- 165 Undercarriage attachment fairing
- 171 Bulkhead section
- 172 Cargo deck
- 173 Soundproofing/insulation lining
- 174 Cabin air duct
- 175 Cargo traversing rails
- 176 Cargo sling/hoist
- 177 Midships emergency rescue cut-out points
- 178 Alt fuselage underfloor fuel tanks (five tanks) capacity 6,375 US gal (24 163 l)
- 179 Cabin side window
- 180 Bulkhead stations
- 181 Wingroot fairing
- 182 Fuselage skinning
- 183 Alt escape hatch (starboard only)
- 184 Alt fuselage structure
- 185 Conditioned air connection (Arctic capability)
- 186 Boom operator's air duct (window demat)
- 187 Alt emergency rescue cut-out points
- 188 Cargo deck opening (boom operator's position access)
- 189 Boom operator's station
- 190 Boom hoist/telescope controls
- 191 Student's/observer's pallet (starboard)
- 192 Boom operator's pallet (port)
- 193 Instructor's pallet (port)
- 194 Scanning side window
- 195 Boom operator's pressure window
- 196 Inward-opening fairing door (wing window)
- 197 Pod pivot/attachment boom
- 198 Cargo deck
- 199 Alt pressure bulkhead
- 200 Fin root fairing
- 201 Fanning formers
- 202 AN/APN-69 receiver transmitter
- 203 Tailfin spar attachment points
- 204 Auto pilot servo
- 205 Alt upper deck tank capacity 2,174 US gal (8 230 l)
- 206 Tank support beam
- 207 Pod line
- 208 Aerial refuelling boom (stowed)
- 209 Tailplane gear screw
- 210 Tailplane gear screw
- 211 Tailplane support fitting
- 212 Tailplane centre-section
- 213 Tailplane end plate
- 214 Ribs
- 215 Port tailplane
- 216 Elevator torque tube
- 217 Elevator internal balance panel
- 218 Port elevator
- 219 Elevator control tab
- 220 Tailplane actuator tab
- 221 Boom outflow doors
- 222 Boom neck fairing
- 223 Tail navigation light
- 224 Crash locator beacon
- 225 Alt frame
- 226 Tab hinge fairing
- 227 Rudder trim tab
- 228 Starboard elevator
- 229 Rudder anti-balance tab
- 230 Fairing
- 231 Rudder frame
- 232 Rudder post
- 233 Rudder internal balance panel
- 234 Rudder control quadrant
- 235 AN/APN-69 radar beacon antenna
- 236 VOR antenna
- 237 Fin leading-edge
- 238 Fin structure
- 239 Rudder fixed trailing edge
- 240 Rudder upper hinge
- 241 HF antennae
- 242 HF probe antenna
- 243 Aerial refuelling boom (deployed)
- 244 Boom operator's window
- 245 Scanning side window

**Boeing KC-135A Stratotanker Specification**

**Power Plant:** Four Pratt & Whitney J57-P-59W turbojets each rated at 13,750 lb st (6,231 kgp) for take-off, with water injection. Fuel capacity, 31,200 US gal (118 105 l) in integral tanks in each wing, bladder tanks in centre section, fore and aft underfloor in the fuselage and above deck in the rear fuselage.

**Performance:** Typical cruising speed, max load refuelling mission, 532 mph (856 km/h) at 10,000 ft; initial rate of climb, 1,290 ft/min (6,65 m/sec); one-engine-out climb rate, 580 ft/min (29.5 m/sec); time to 34,500 ft (9 300 m), 27 min; critical field length, max weight, ISA +17 deg C at sea level, 13,700 ft (4 176 m); mission radius, 3,000 naut miles (5 552 km) to offload 24,000 lb (10 886 kg) fuel, 2,000 naut miles (3 700 km) to offload 74,000 lb (33 566 kg) fuel, 1,000 naut miles (1 850 km) to offload 120,000 lb (54 432 kg) fuel.

**Weights:** Operating weight empty, 106,306 lb (48 230 kg); normal take-off weight, 301,000 lb (136 806 kg); max overload weight, 316,000 lb (143 338 kg).

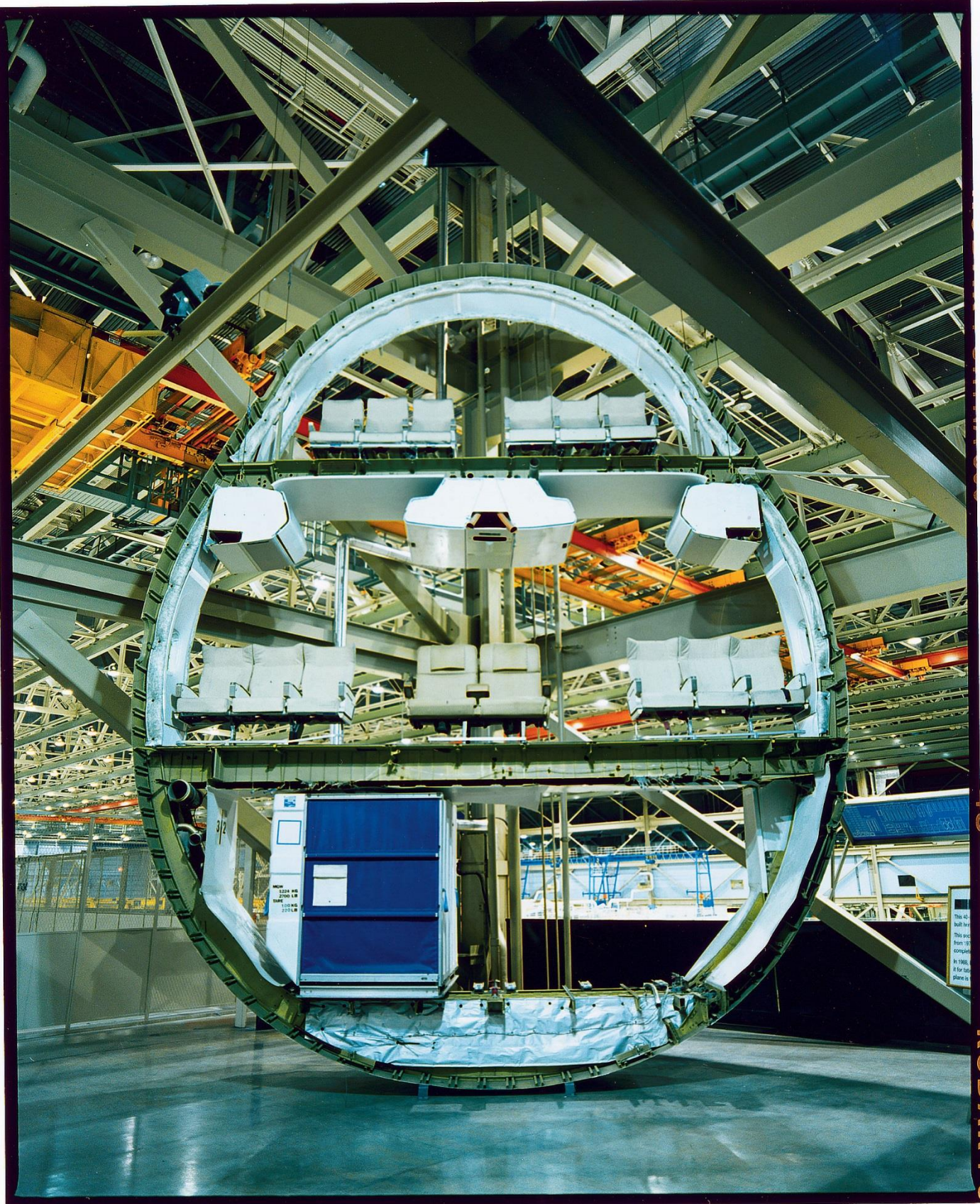
**Dimensions:** Span, 130 ft 10 in (39.88 m); wing length, 134 ft 6 in (40.99 m); height overall, 41 ft 8 in (12.69 m); length area, 2,433 sq ft (226.03 m<sup>2</sup>); sweepback, 35 deg on quarter chord line; dihedral, 7 deg constant; undercarriage track, 22 ft 1 in (6.73 m); wheelbase, 45 ft 8 in (13.92 m).

**PILOT PRESS COPYRIGHT DRAWING**

- 120 Pylon/wing fairing
- 121 Pylon strut access
- 122 No 2 engine pylon
- 123 Nacelle
- 124 Engine inlet nose cowl
- 125 Oil cooler chin intake scoop Pratt & Whitney J57-P-59W
- 126 Oil tank
- 127 Engine drive alternators
- 128 Oil tank
- 129 Tail pipe
- 130 Exhaust outlet cone
- 131 Nacelle support aft fairing
- 132 Leading-edge wing flap

- 204 Auto pilot servo
- 205 Alt upper deck tank capacity 2,174 US gal (8 230 l)
- 206 Tank support beam
- 207 Pod line
- 208 Aerial refuelling boom (stowed)
- 209 Tailplane gear screw
- 243 Aerial refuelling boom (deployed)
- 244 Boom operator's window
- 245 Scanning side window
- 246 Aerial refuelling boom (five)
- 247 Air-conditioning pack intake scoop
- 248 Cargo door (open)
- 249 Underfloor forward fuel tanks (four)
- 250 Refuelling envelope central director

# Boeing 747 Fuselage Cross-Section





# Composite Aircraft







# GULFSTREAM G250

MEGGITT

Rockwell Collins

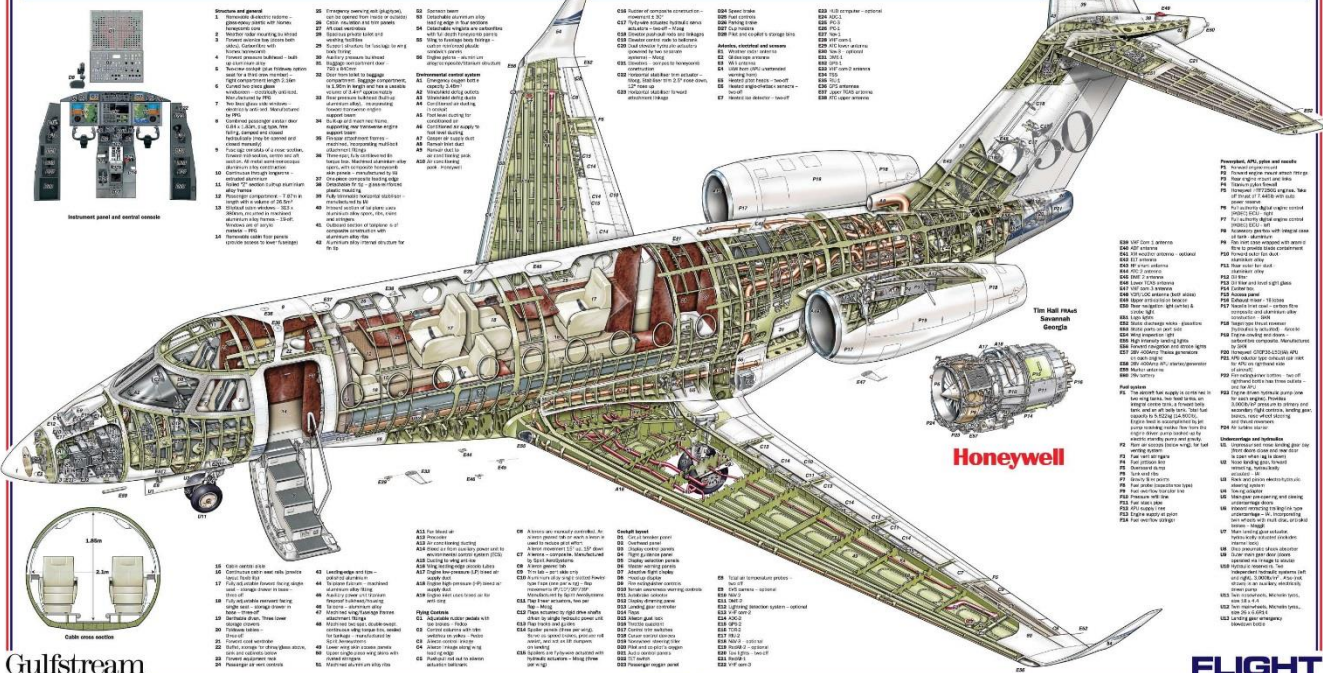
MOOG

PLG

PPG Aerospace Transparencies

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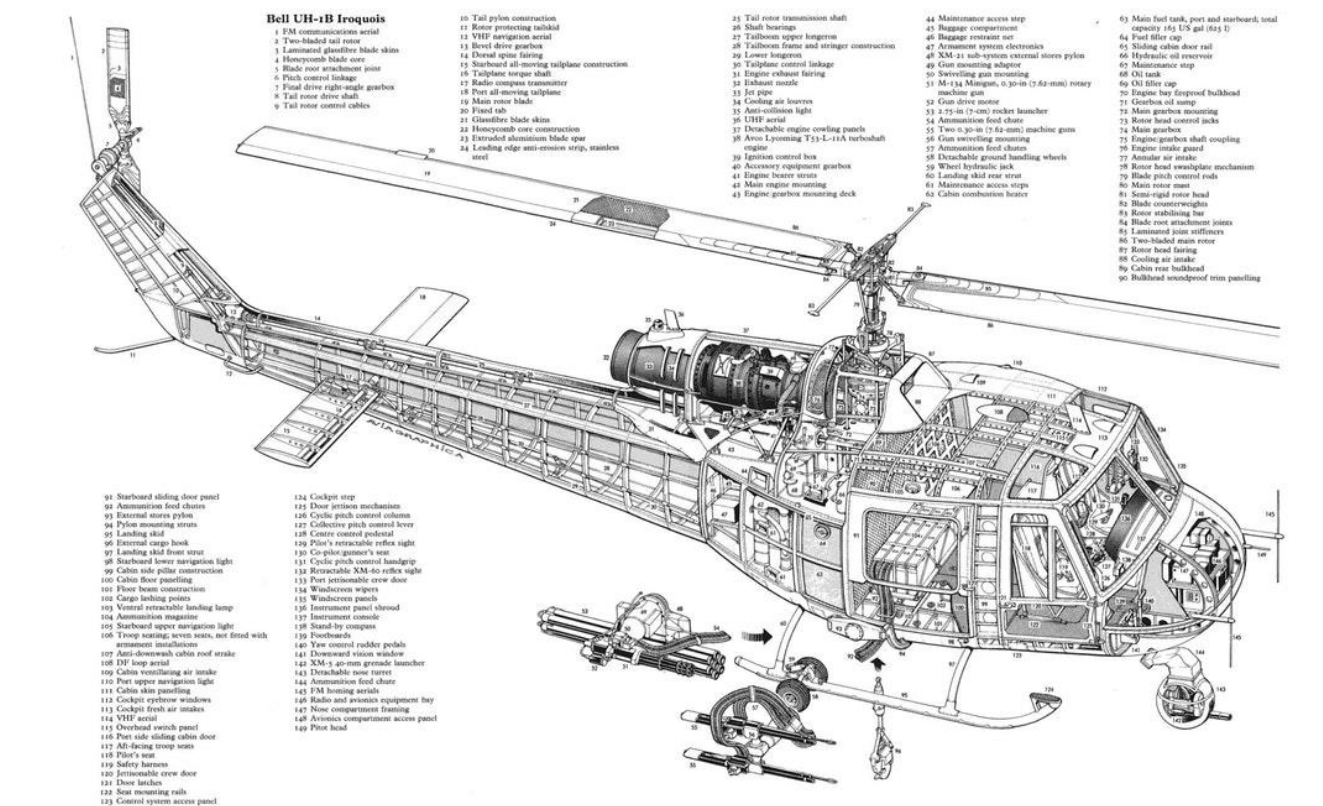
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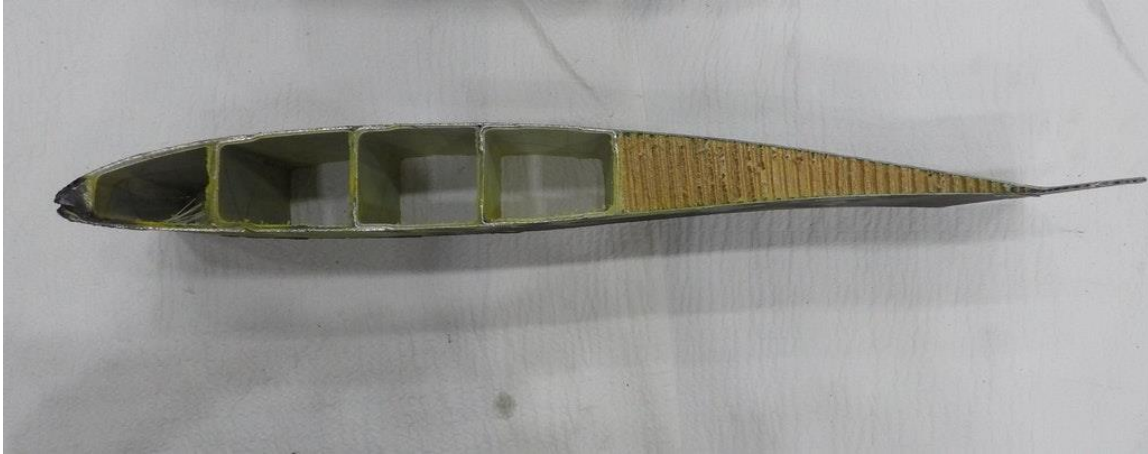


124 Cockpit strip  
125 Door jettison mechanism  
126 Cyclic pitch control column  
127 Collective pitch control lever  
128 Control control pedestal  
129 Pilot's retractable reflex sight  
130 Co-pilot gunner's seat  
131 Cyclic pitch control handgrip  
132 Retractable XM-10 reflex sight  
133 Port jettisonable crew door  
134 Window covers  
135 Window covers  
136 Fuel tank  
137 Stand-by compass  
138 Fuel tank  
139 Fuel tank  
140 Yaw control rudder pedals  
141 Fuel tank  
142 XM-140-gm grenade launcher  
143 Fuel tank  
144 Annunciation feed chart  
145 FM housing aerials  
146 Radio and avionics equipment bay  
147 Nose compartment framing  
148 Avionics compartment access panel  
149 Pilot head

## Bell UH-1B Iroquois

- 10 Tail pylon construction
- 11 Rotor protecting tubular
- 12 VHF navigation aerial
- 13 Dorsal spine fairing
- 14 Blade drive gearbox
- 15 Tail rotor torque shaft
- 16 Radio compass transmitter
- 17 Fuel tank
- 18 Main rotor blade
- 19 Main rotor blade
- 20 Blade drive gearbox
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# Apache Helicopter Blade



# Westland Helicopter Blade

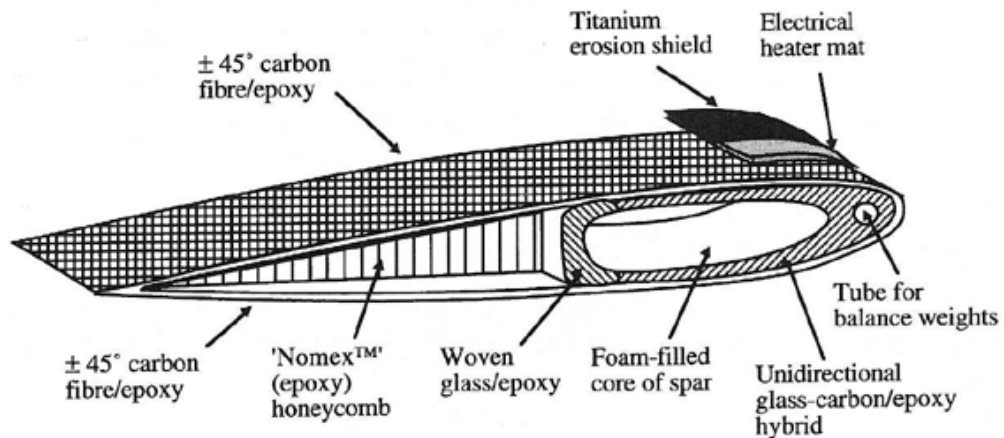
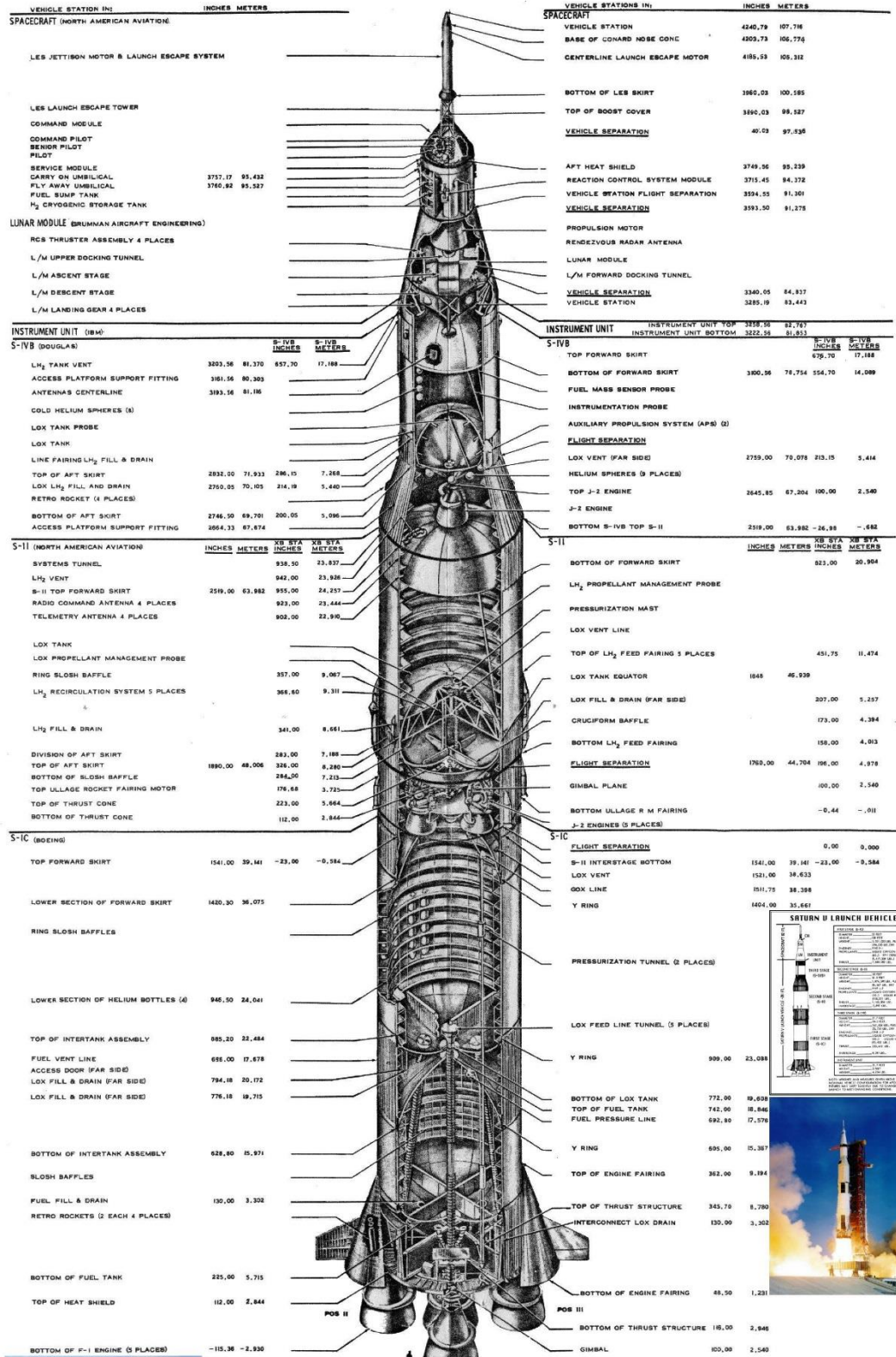


Fig. 12.4 Schematic section through a typical composite construction for a helicopter rotor blade. (Courtesy of Westland Helicopters.)

# Saturn V + Apollo = $6.5 \times 10^6$ lbs



NOTE: S-IC STAGE ROTATED 45° COUNTER CLOCKWISE FOR CLARITY

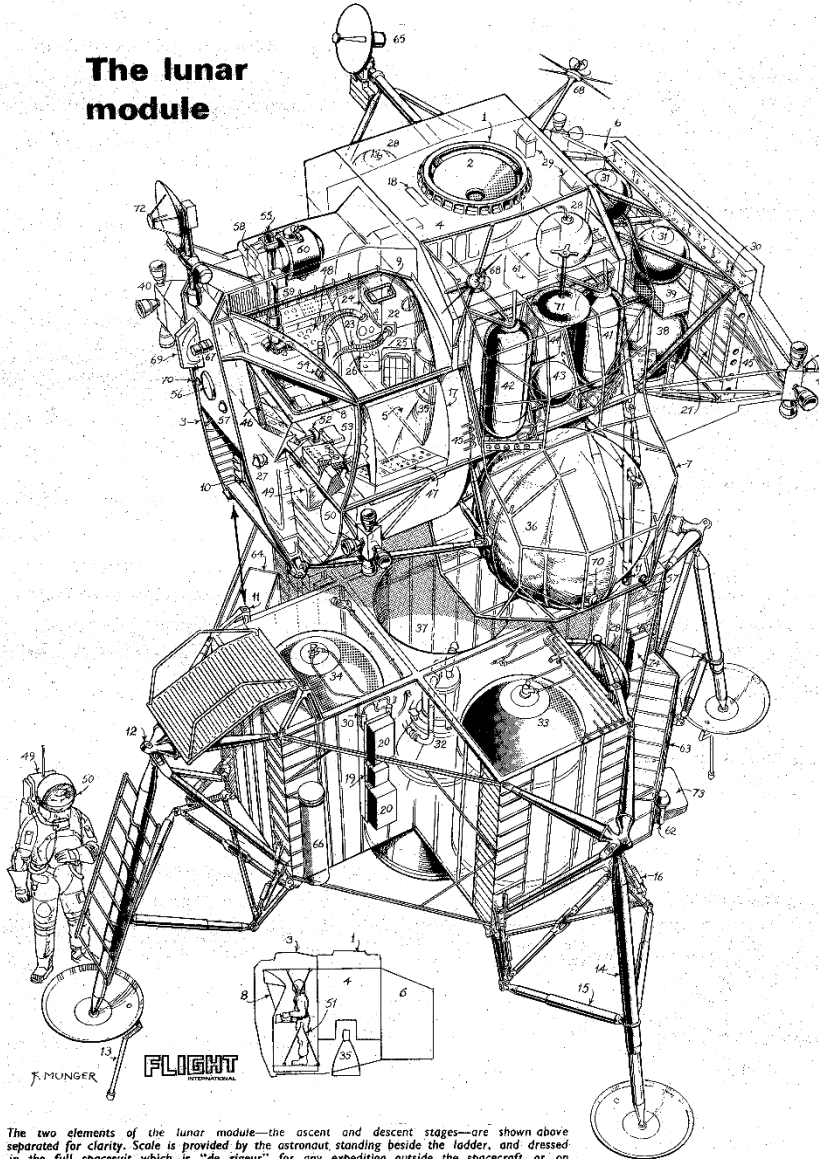


# Lunar Module

-Ascent: 4700 lbs dry  
10000 lbf gross

-Descent: 23000 lbs

**The lunar module**



The two elements of the lunar module—the ascent and descent stages—are shown above separated for clarity. Scale is provided by the astronaut standing beside the ladder, and dressed in the full spacesuit which is “de rigueur” for any expedition outside the spacecraft or on the Moon. The descent stage provides a launch platform for the ascent stage. At lift-off, the upper-stage engine exhausts through the central tunnel in the lower stage.

# Orion Capsule



By U.S. Navy -



