

FAQS for HW1

Units:

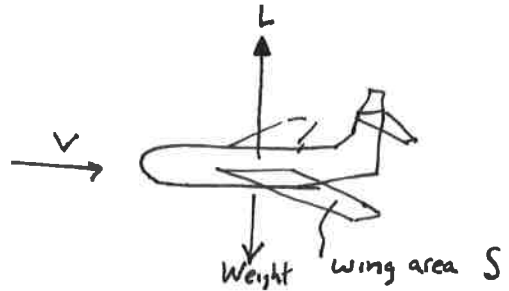
$$\rho = \frac{PM}{RT} = 1 \frac{\text{lb}_f}{\text{in}^3} \frac{28.97 \text{ lb}_m}{\text{lb}_m} \frac{R \text{ lb}_m}{1545.34 \text{ ft lb}_f} \frac{\text{slug}}{32.174 \text{ lb}_m} \frac{144 \text{ in}^2}{\text{ft}^2}$$

$\underbrace{\hspace{10em}}_{\bar{R} \text{ is universal gas constant}} \quad \underbrace{\hspace{10em}}_{\text{If your density is } \approx 5 \text{ times too low, here is the problem!}}$

$$= [\text{slug}/\text{ft}^3]$$

Aircraft Lift

$$L = \frac{1}{2} \rho V^2 C_L S$$



For an identical aircraft

$$\frac{1}{2} \rho_1 V_1^2 C_L S = W = L = \frac{1}{2} \rho_2 V_2^2 C_{L_{max}} S$$

$\underbrace{\hspace{15em}}_{\text{same}}$

Thus

$$\rho_1 V_1^2 = \rho_2 V_2^2 \Rightarrow \frac{V_2}{V_1} = \sqrt{\frac{\rho_1}{\rho_2}}$$

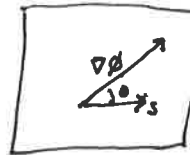
Velocity ratio scales with $\sqrt{\rho_1/\rho_2}$

Directional Derivative

$$\nabla \phi \cdot S = \nabla_S \phi = (S \cdot \nabla) \phi =$$

$$\uparrow = |\nabla \phi| |S| \cos \theta$$

\uparrow this needs to be 1, since S is a unit vector



Particle in tube

$x = \int u dt$, so you could find $\frac{DT}{Dt}$ with calculus and substitution in 1D.

Or, be generic and use the substantial derivative.

$$\frac{DT}{Dt} = \frac{dT}{dt} + V \cdot \nabla T$$