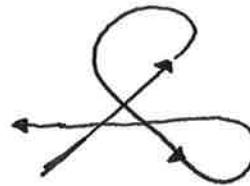


Cloverleaf

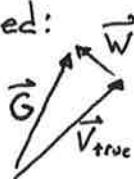
Airspeed calibration for high speed aircraft with no reference/visual to the ground and arbitrary airspeeds + headings.



We have 3 unknowns:

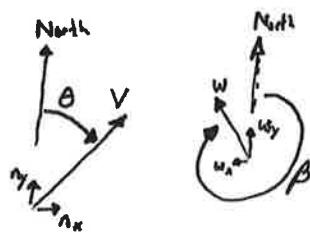
- KIAS \rightarrow KCAS "correction"
- Wind speed
- Wind direction

Ground speed:



$$\vec{G} = \vec{V}_{\text{true}} + \vec{W} = |V_{\text{true}}| (n_x \ n_y) + |W| (w_x \ w_y) = |G| (g_x \ g_y)$$

Vector components in N and E directions



Absolute value

$$|G|^2 = \underbrace{(|V_{\text{true}}| n_x + |W| w_x)^2}_{x\text{-component}} + \underbrace{(|V_{\text{true}}| n_y + |W| w_y)^2}_{y\text{-component}}$$

$$= (V n_x + W w_x)^2 + (V n_y + W w_y)^2$$

$$= \underbrace{\sqrt{n_x^2 + w_x^2}^2 + \underbrace{\sqrt{n_y^2 + w_y^2}^2}_{w_x^2 + w_y^2 = 1} + 2VW n_x w_x + 2VW n_y w_y}$$

$$\begin{aligned} n_x &= \sin \theta \\ n_y &= \cos \theta \\ w_x &= \sin \beta (-1) \\ w_y &= \cos \beta (-1) \end{aligned}$$

wind direction
is "from" not
towards.

Airspeed:

$$V_{\text{true}} = f(V_{\text{cas}})$$

$$= f(V_{\text{ias}} + \text{correction})$$

$$G^2 = V^2 + W^2 + 2VW(n_x w_x + n_y w_y) = V^2 + W^2 + 2VW(\sin \theta \sin \beta + \cos \theta \cos \beta)$$

$$G^2 = f(V_{\text{ias}} + c)^2 + W^2 - 2f(V_{\text{ias}} + c)W(\sin \theta \sin \beta + \cos \theta \cos \beta)$$

this has to be true for any airspeed, direction, wind, etc.

GPS:

From a GPS, we can find groundspeed. We can read off Vias and heading θ and terms to calc $f()$

Solution:

$$G^2 = f(V_{\text{ias}} + c)^2 + W^2 - 2f(V_{\text{ias}} + c)W(\sin \theta \sin \beta + \cos \theta \cos \beta)$$

Known Known Unknown Unknown Known Unknown
 ↓ ↑ ↑ ↑ ↓ ↑
 ① ② ③ ④ ⑤ ⑥

3 unknowns so we need 3 equations (i.e. 3 independent lines)
 A cloverleaf!

Numerical solution (one possibility) :

$$\begin{pmatrix} U(x, y, z) \\ V(x, y, z) \\ W(x, y, z) \end{pmatrix} = \begin{pmatrix} U_0 \\ V_0 \\ W_0 \end{pmatrix} + \underbrace{\begin{bmatrix} \frac{dU}{dx} & \frac{dU}{dy} & \frac{dU}{dz} \\ \frac{dV}{dx} & \frac{dV}{dy} & \frac{dV}{dz} \\ \frac{dW}{dx} & \frac{dW}{dy} & \frac{dW}{dz} \end{bmatrix}}_{\text{Jacobian } J} \begin{pmatrix} \Delta x \\ \Delta y \\ \Delta z \end{pmatrix} + \text{higher order terms}$$

Jacobian = J

If we set the LHS to zero, we can solve for the $\Delta x, \Delta y, \Delta z$ that makes this true

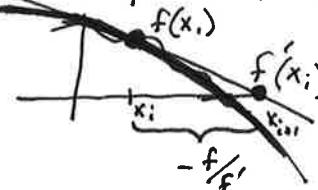
$$0 = \begin{pmatrix} U_0 \\ V_0 \\ W_0 \end{pmatrix} + [J] \begin{pmatrix} \Delta x \\ \Delta y \\ \Delta z \end{pmatrix} \Rightarrow \begin{pmatrix} \Delta x \\ \Delta y \\ \Delta z \end{pmatrix} = -[J]^{-1} \begin{pmatrix} U_0 \\ V_0 \\ W_0 \end{pmatrix}$$

Then correct the x, y, z guess by $\Delta x, \Delta y, \Delta z$:

$$\begin{pmatrix} x \\ y \\ z \end{pmatrix}_{i+1} = \begin{pmatrix} x \\ y \\ z \end{pmatrix}_i + \begin{pmatrix} \Delta x \\ \Delta y \\ \Delta z \end{pmatrix} = \begin{pmatrix} x \\ y \\ z \end{pmatrix}_i - [J]^{-1} \begin{pmatrix} U_0(x_i, y_i, z_i) \\ V_0 \\ W_0 \end{pmatrix}$$

This is the multidimensional equivalent of Newton-Raphson's method

$$x_{i+1} = x_i - \frac{f(x_i)}{f'(x_i)}$$



Really fast convergence to zero.... when it works...

Residual:

$$R = f(V_{bias} + c)^2 + W^2 - 2 f(V_{bias} + c) W (\sin \theta \sin \beta + \cos \theta \cos \beta)$$

States are: W, β, c

$\frac{dR}{dc}$ is NOT too hard to find, but $\frac{dR}{d\beta}$ is really not analytically possible. $\frac{dR(f(V_{bias} + c)^2)}{dc} = ?$

We can use numerical derivatives $\frac{dR}{dc} \approx \frac{R(c + \delta c) - R(c - \delta c)}{2\delta c}$ where δc is a small const $\approx 0.001 \sim 50$

$$\begin{pmatrix} R_{line 1} \\ R_{line 2} \\ R_{line 3} \end{pmatrix} \rightarrow \begin{pmatrix} U_0 \\ V_0 \\ W_0 \end{pmatrix} \text{ above derivatives}$$

$$\text{and } \begin{bmatrix} \frac{dR_1}{dw} & \frac{dR_1}{d\beta} & \frac{dR_1}{dc} \\ \frac{dR_2}{dw} & \ddots & \vdots \\ \frac{dR_3}{dw} & \ddots & \frac{dR_3}{dc} \end{bmatrix} = J$$

Iterate:

Iterate several times to find W, β, c

Cloverleaf Ground Track

"use ground track rather than heading"



$$|G|(g_x \ g_y) = |\mathbf{V}|(n_x \ n_y) + |\mathbf{W}|(w_x \ w_y)$$

solve for V later

$$|\mathbf{V}|(n_x \ n_y) = |G|(g_x \ g_y) - |\mathbf{W}|(w_x \ w_y) = (G_{gx} - W_{wx}, G_{gy} - W_{wy})$$

Square

$$V^2 = \underbrace{G^2 g_x^2 + W^2 w_x^2 - 2GWg_x w_x}_{\text{---}} + \underbrace{G^2 g_y^2 + W^2 w_y^2 - 2GWg_y w_y}_{\text{---}}$$

$$= G^2 + W^2 - 2GW(g_x w_x + g_y w_y)$$

so the residual equation is

$$\begin{aligned} R &= G^2 + W^2 - V^2 - 2GW(g_x w_x + g_y w_y) \\ &= G^2 + W^2 - f(V_{bias} + c)^2 + 2GW(\sin\phi \sin\beta + \cos\phi \cos\beta) \end{aligned}$$

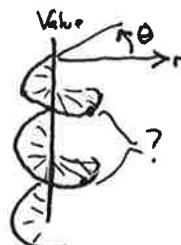
Why might this be better?

- GPS gives ground track to deg
- True heading is not so simple (magnetic declination, instrument bias, drift, wind resolution)

Huge advantage

However, this implementation does not work well! Why?

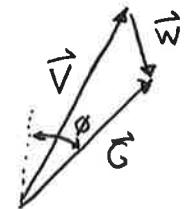
- The wind vector is the culprit. When $W=0$, the Jacobian is rank deficient (i.e. not invertible)
- $\frac{d(xy)}{dx} = y$, but if $y=0$, then $\frac{d(xy)}{dx}$ is zero for any x . Optimize?
- The math reason is that we are saying that $|W|$ and β are independent, but this is not true
If the wind is 270° at 2^{kt} and then $\Delta kt = -3^{kt}$, what is the wind vector?
Is this an injective function?



Avoid using this polar representation

Cloverleaf Generic w GPS

$$|G|(g_x \ g_y) = \underbrace{|V|(n_x \ n_y)}_{\text{ground speed vector}} + \underbrace{|W|(w_x \ w_y)}_{\text{wind vector}}$$



Using a GPS:

We know $|G|$, track and airspeed, the unknowns are wind velocity and direction.

However, for math reasons, we don't want to use wind direction, but rather wind components.

We also don't want to use the true airspeed direction (i.e. less precision than actual gd track).

$$|V|(n_x \ n_y) = |G|(g_x \ g_y) - \underbrace{|W|(w_x \ w_y)}_{\text{written as } (w_x \ w_y)}$$

Find magnitude:

$$V^2 = (G g_x - w_x)^2 + (G g_y - w_y)^2$$

Find a residual function to drive to zero.

$$R = (G g_x - w_x)^2 + (G g_y - w_y)^2 - V^2$$

Add Substitute for ground track direction,

$$g_x = \sin \phi \quad g_y = \cos \phi$$

$$R = (G \sin \phi - w_x)^2 + (G \cos \phi - w_y)^2 - V^2$$

Substitute from IAS to TAS

$$R = (G \sin \phi - w_x)^2 + (G \cos \phi - w_y)^2 - f(\overset{\text{IAS} + C}{\text{TAS}}, \text{alt, temp})^2$$

Numerical:

Newton Raphson iteration

$$\vec{U} = \begin{pmatrix} U(x, y, \dots) \\ V(x, y, \dots) \end{pmatrix} = \begin{pmatrix} U_0 \\ V_0 \end{pmatrix} + \begin{bmatrix} \frac{dU}{dx} & \frac{dU}{dy} & \frac{dU}{d\dots} \\ \frac{dV}{dx} & \ddots & \ddots \\ \vdots & \ddots & \ddots \end{bmatrix} \begin{pmatrix} \Delta x \\ \Delta y \\ \Delta z \end{pmatrix} + \text{higher order terms}$$

Solving for $\vec{U} = 0$

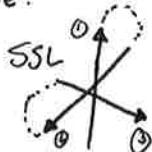
$$\begin{pmatrix} \Delta x \\ \Delta y \\ \vdots \end{pmatrix} = -[\vec{J}]^{-1} \begin{pmatrix} U_0 \\ V_0 \\ \vdots \end{pmatrix}$$

$$\begin{pmatrix} w_x \\ w_y \\ C \end{pmatrix} = -[\vec{J}] \begin{pmatrix} w_x \\ w_y \\ C \end{pmatrix}$$

$$\text{and } \frac{dU}{dx} \approx \frac{U(x+\delta) - U(x-\delta)}{2\delta} \text{ with } \delta \ll$$

Repeat 10-100 times or set a terminating criterion

Example:



- ① Heading $\approx N$, Track 10° , 112 kt ground speed, IAS is 115 kts
- ③ Heading $\approx SE$, Track 116° , 128 kt ground speed, 115 kt IAS \Rightarrow Wind 270 at 20, C is -5 kts
- ② Heading $\approx NW$, Track 233° , 88 kt ground speed, 110 kt IAS

Example 2:

Heading	KIAS	GS	Alt	OAT
358.2	105	162.8	14500	0
113.6	102	127.3	14600	5
248.4	100	132.5	14200	2



Recoveries: Wind 165 at 20, C is -5 kts